

SSME ALTERNATE TURBOPUMP DEVELOPMENT PROGRAM (HPFTP)

VERIFICATION COMPLETE REPORT FIRST TURBINE BLADE AERODYNAMIC DESIGN DVS DR NO. 3.1.2.2.4.1, VM NO. 4.1.2.4 A

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Marshall Space Flight Center, AL 35812

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(NASA-CR-183751) SSME ALTERNATE TURBOPUMP
DEVELOPMENT PROGRAM (HPFTP). VERIFICATION
COMPLETE REPORT. FIRST TURBINE BLADE
AERODYNAMIC DESIGN DVS DR NO. 3.1.2.2.4.1,
VM NO. 4.1.2.4 A (Pratt and Whitney)

N89-71482

00/20 Unclassified
 0233362



**UNITED
TECHNOLOGIES
PRATT&WHITNEY**

HPFTP Turbine Aerodynamic Design

The High Pressure Fuel Turbopump (HPFTP) turbine aerodynamic design is based on the requirements defined by the Interface Control Document (ICD) and by the Power Balance Model, Table 387B. Performance Table 387B was used for the turbine aerodynamic design because its turbine flow capacities are consistent with the baseline turbine nozzle flow test results conducted on Pratt & Whitney's test stand, E-6, in December, 1986.

A conventional pressure-compounded, 2-stage turbine was chosen because of its inherent high efficiency over a wide range of steady-state operating conditions. No exit guide vane is required for the small (18 degrees) exit swirl angle. The high airfoil gas bending loads in the HPFTP turbine required thin wall, hollow airfoil sections, with larger moments of inertia, in order to reduce the airfoil bending stresses. The HPFTP turbine design has a mean diameter wheel speed of 1482 ft/sec which is compatible with allowable disk and root attachment stress criteria. This wheel speed also provides a high design point wheel speed to gas velocity ratio, assuring that there will not be a significant efficiency loss at minimum power level (MPL) operation. The design point velocity ratio, (0.55) is conservative, ensuring minimal aerodynamic risk. The design speed of approximately 36,500 rpm selected for the HPFTP was primarily set by the pump hydrodynamics. The height of the turbine annulus was selected to limit the last stage blade root centrifugal stress to 46,000 psi. This annulus size yielded a favorable exit Mach number of 0.18 and a low exit swirl angle of 18 degrees, therefore, this rpm was satisfactory to the performance, stress, and exit Mach number requirements of the turbine.

The methodology associated with the design of the HPFTP starts with the meanline design analysis. This analysis is based on the assumption that the flow through the turbine can be represented by the flow at the center of the flow passage. This simplified approach permits selection of the number of stages required, the mean diameter of the flow passage, and the annulus area. Included in the analysis is an estimate of the aerodynamic efficiency. This prediction system uses the physical laws of aerodynamics and correlations from rig and engine data to estimate profile loss, secondary loss, blade tip leakage, and shock and incidence losses based on the geometry and aerodynamic parameters of the turbine. An interactive graphic flowpath design system is used, in conjunction with the optimum meanline design, to generate candidate flowpath configurations.

The streamline design analysis is used to optimize the radial variation in the velocity triangles, once the average conditions are selected from the meanline analysis. This analysis calculates the flow characteristics at numerous radial locations and at the inlet and exit of each airfoil row. Once the meanline and streamline analyses have been used to optimize the velocity triangles throughout the turbine, 2 dimensional (2-D) airfoil sections are designed. These airfoil sections are designed to achieve contours that provide the desired amount of flow turning without permitting the flow to separate from

the airfoil surface. This process involves determining the static pressure distributions and boundary layer parameters along the airfoil surfaces and endwalls. An interactive graphics airfoil design system is used to identify adverse static pressure gradients such that the airfoil contour can be modified appropriately. After the 2-D airfoils are estimated at several spanwise locations, they are radially faired and combined with a preliminary endwall definition. An inviscid multi-stage 3-D flow analysis is then used to refine and optimize the entire flowpath configuration.

All turbine airfoil, endwall, inlet, and exit flow passage surfaces are contoured and refined as a system. The multi-stage feature enables a complete evaluation of potential changes to an individual surface contour during the design process. This assessment includes, not only flow property changes around the component being modified, but also around all upstream and downstream components in the complete turbine system. Improved performance and reduced risk result from this global optimization capability.

This report contains:

- o Hot elevation diagrams for each airfoil
- o 3-D airfoil plots
- o 2-D airfoil section plots
- o Tabulated airfoil section coordinates
- o A plot of hot gaging dimensions versus radius
- o A plot of percent change in flow area versus airfoil rotation
- o A plot of stress versus span
- o 3-D airfoil static pressure distributions
- o Airfoil Ps/PT and Mach number contours
- o A plot of suction surface boundary layer friction coefficient versus surface distance

COVER SHEET

S.S.M.E.
ENGINE Alternate Turbo-Pump Development
AIRFOIL 1st Stage Blade

ENGINEER R.J. Rowey EXT 5962 DATE 8/24/87

AERODYNAMIC DESIGN POINT 109% Power - Design Table 0387.B dated 4/10/87

F.T.D. LIST:

ELEVATION _____

AIRFOIL SECTIONS _____

AIRFOIL COORDINATES _____

DF LIST:

GAGING VS. RADIUS _____

FLOW AREA VS. ROTATION _____

STRESS VS. % SPAN _____

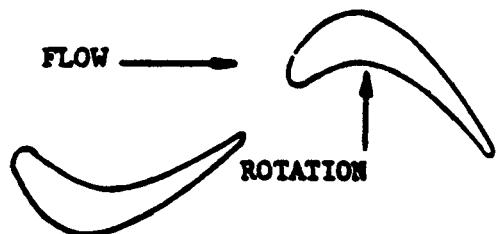
PRESSURE DIST. _____

BOUNDARY LAYER _____

CHECK ONE

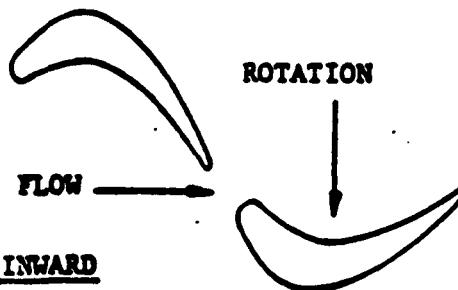
P&WA CONVENTIONAL ROTATION

VANE _____ BLADE



P&WA COUNTER ROTATION

VANE _____ BLADE _____



VIEW LOOKING RADIALLY INWARD

SSME HPTIP FIRST STAGE BLADE
HOT ELEVATION

0.1692

— 0.500 —
AXIAL CHORD

TIP DEFINING SECTION

5.085 R.

1/4 TIP DEFINING SECTION

4.835 R.

MEAN DEFINING SECTION

4.6025 R.

1/4 ROOT DEFINING SECTION

4.36125 R.

ROOT DEFINING SECTION

— 0.900 —
AXIAL CHORD

4.12 R.

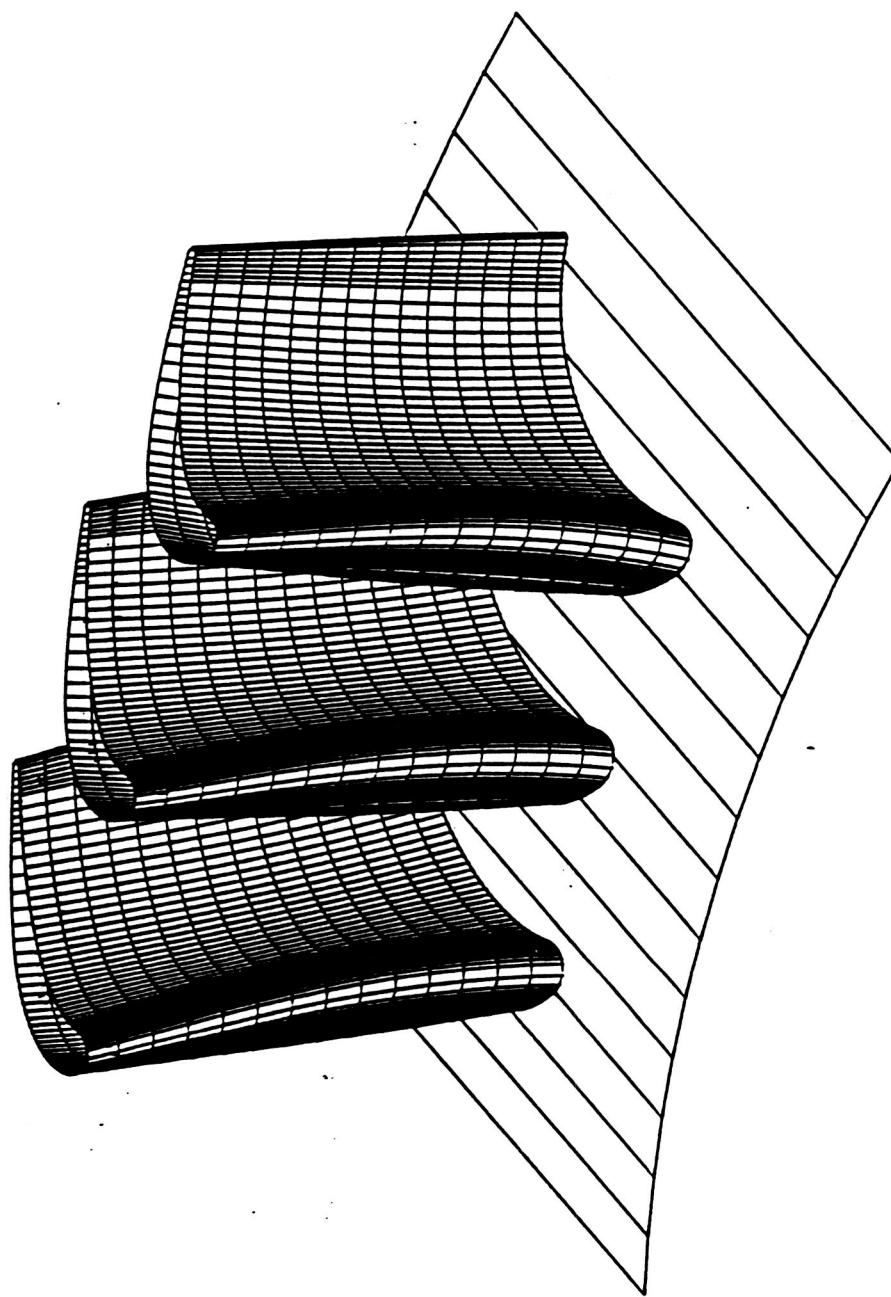
— 0.169 —
RADIAL REFERENCE LINE

(X,Y) = 0.0, 0.0

- NOTES:
1. BLADE IS CURVED LINE FAIRED IN PR24.
 2. NUMBER OF SPLINES IS 30.
 3. ALL DIMENSIONS ARE HOT.
 4. MATERIAL IS ALUMINUM DENSITY = 0.317 #/IN.³
 5. NET WEIGHT IS 0.0369 #/FEOD.
 6. POLE IS 36% DEAD AT 36420 RPM.
 7. AIRFOIL COORDINATES RESIDE IN PR24 FILE
RERPR24.BEFINAL.

R.J.P. 8/20/87

3D PLOT



SSME FT FINAL 1B... R.J. ROWEY... 5-14-87..

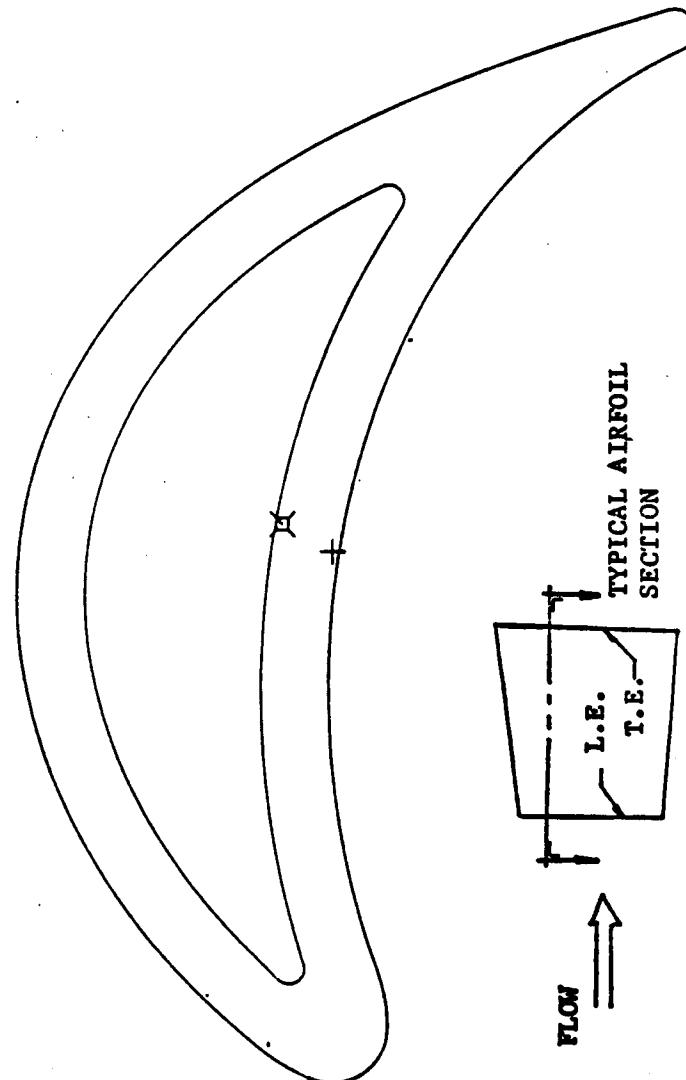
05/22/87 14:22:31
40.89 48.59 40.89

SSME FT FINAL IRB... R.J. ROWLEY 6-25-87.

ROOT

CYLINDRICAL
SCALE 10.0
THERMAL SHRINK FACTOR 1.00000
08/19/87
11:42:38

NUMBER OF BLADES	50.
RADIUS (HOT)	4.120 INCHES
CRACING (HOT)	0.4892 INCHES
PITCH (HOT)	0.5177 INCHES
AER. WIDTH	0.9000 INCHES
BLADE INLET ANGLE	60.565 DEGREES
CAS INLET ANGLE	44.718 DEGREES
BLADE EXIT ANGLE	21.129 DEGREES
CAS EXIT ANGLE	21.062 DEGREES
CRACING ANGLE	21.431 DEGREES
UNCOVERED TURNING	22.892 DEGREES
LEADING EDGE RADIUS	0.0454 INCHES
TRAILING EDGE RADIUS	0.0175 INCHES
TOTAL AREA (SOLID)	0.1889 SQ. IN.
METAL AREA	0.1137 SQ. IN.
(NET. UNCOATED)	

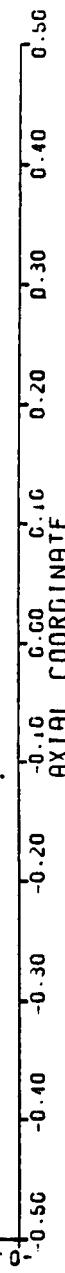


X C.C.
+ STACKING LINE
CRACE

X C.C.
+ STACKING LINE
CRACE

CLOCKWISE ROTATION WHEN LOOKING FORWARD

NOMINAL ENGINE POSITION

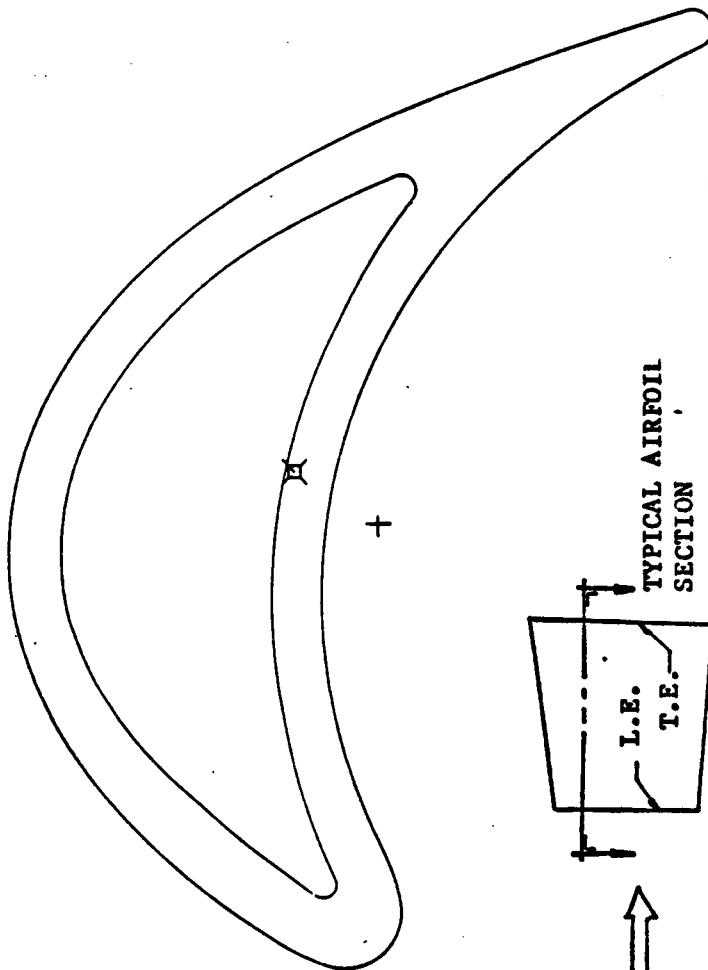


SSME FT FINAL I.B. R.J.RGWY 6-25-87

1/4 ROOT

CYLINDRICAL
SCALE 10.0
THERMAL SHRINK FACTOR 1.00000
08/19/87
11.42:38

NUMBER OF BLADES	50.
RADIUS (HOT)	4.361 INCHES
GAGING (HOT)	0.1970 INCHES
PITCH (HOT)	0.5481 INCHES
AXIAL WIDTH	0.8660 INCHES
BLADE INLET ANGLE	53.33 DEGREES
CAS INLET ANGLE	38.220 DEGREES
BLADE EXIT ANGLE	21.126 DEGREES
CAS EXIT ANGLE	21.062 DEGREES
GAGING ANGLE	21.062 DEGREES
UNCOVERED TURNING	21.153 DEGREES
LEADING EDGE RADIUS	0.0467 INCHES
TRAILING EDGE RADIUS	0.0163 INCHES
TOTAL AREA (SOLID)	0.1655 SQ. IN.
METAL AREA	0.0875 SQ. IN.
(NET, UNCORRECTED)	



ENGINE
CLOCKWISE ROTATION WHEN LOOKING FORWARD
NOMINAL ENGINE POSITION

X C.G. 0.0133 0.0000
+ STACKING LINE 0.0000 0.0000
GAGE 0.2430 0.0000

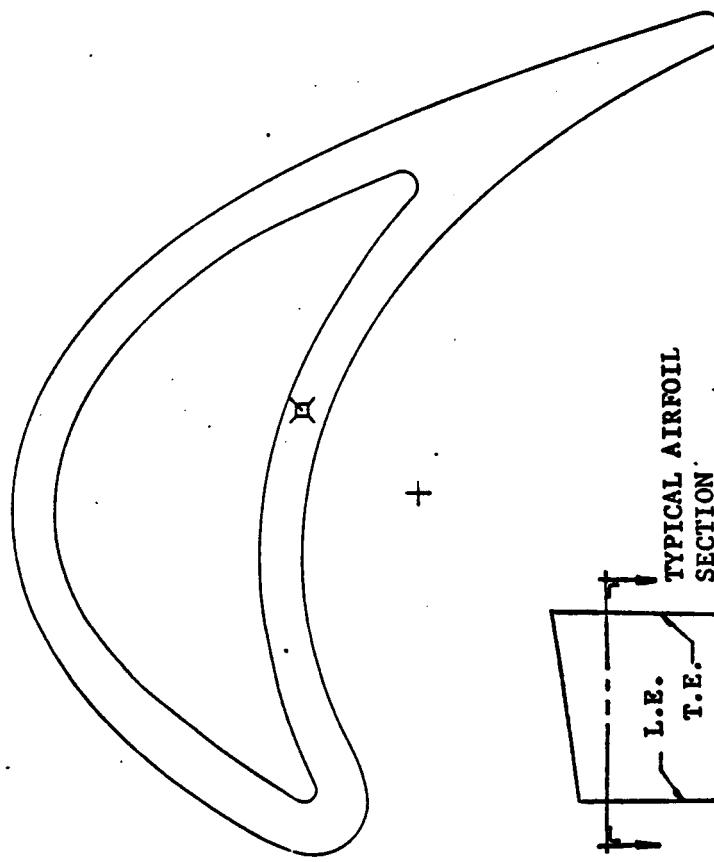


SSME FT FINAL IR. R. J. ROWER 6-25-87

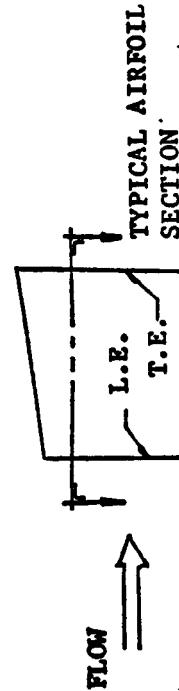
MEAN

CYLINDRICAL
SCALE 10.0
THERMAL SHRINK FACTOR 1.000000
08/19/87
11-42:38

NUMBER OF BLADES	36.
RADIUS (INCH)	4.602 INCHES
CAGING (INCH)	0.2065 INCHES
PITCH (INCH)	0.5784 INCHES
AIRFOIL WIDTH	0.7000 INCHES
BLADE INLET ANGLE	51.693 DEGREES
GAS INLET ANGLE	38.710 DEGREES
BLADE EXIT ANGLE	21.063 DEGREES
GAS EXIT ANGLE	21.062 DEGREES
CAGING ANGLE	20.914 DEGREES
UNCOVERED TURNING	20.619 DEGREES
LEADING EDGE RADIUS	0.0448 INCHES
TRAILING EDGE RADIUS	0.0150 INCHES
TOTAL AREA (SOLID)	0.1353 SQ. IN.
METAL AREA	0.0685 SQ. IN.
(NET UNCOATED)	



+



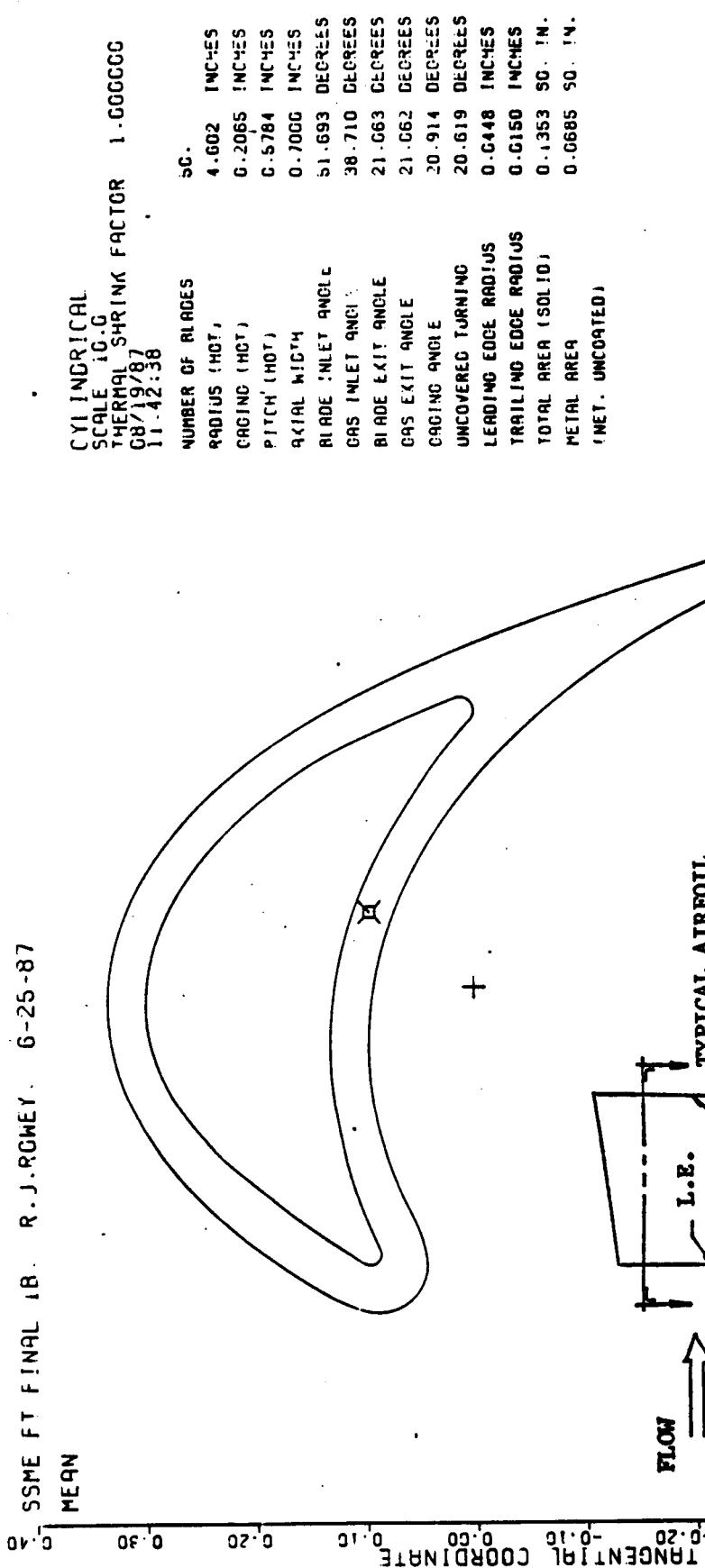
TYPICAL AIRFOIL SECTION



X	Y
0.0000	0.0000
0.0000	0.0000
0.2065	0.0690

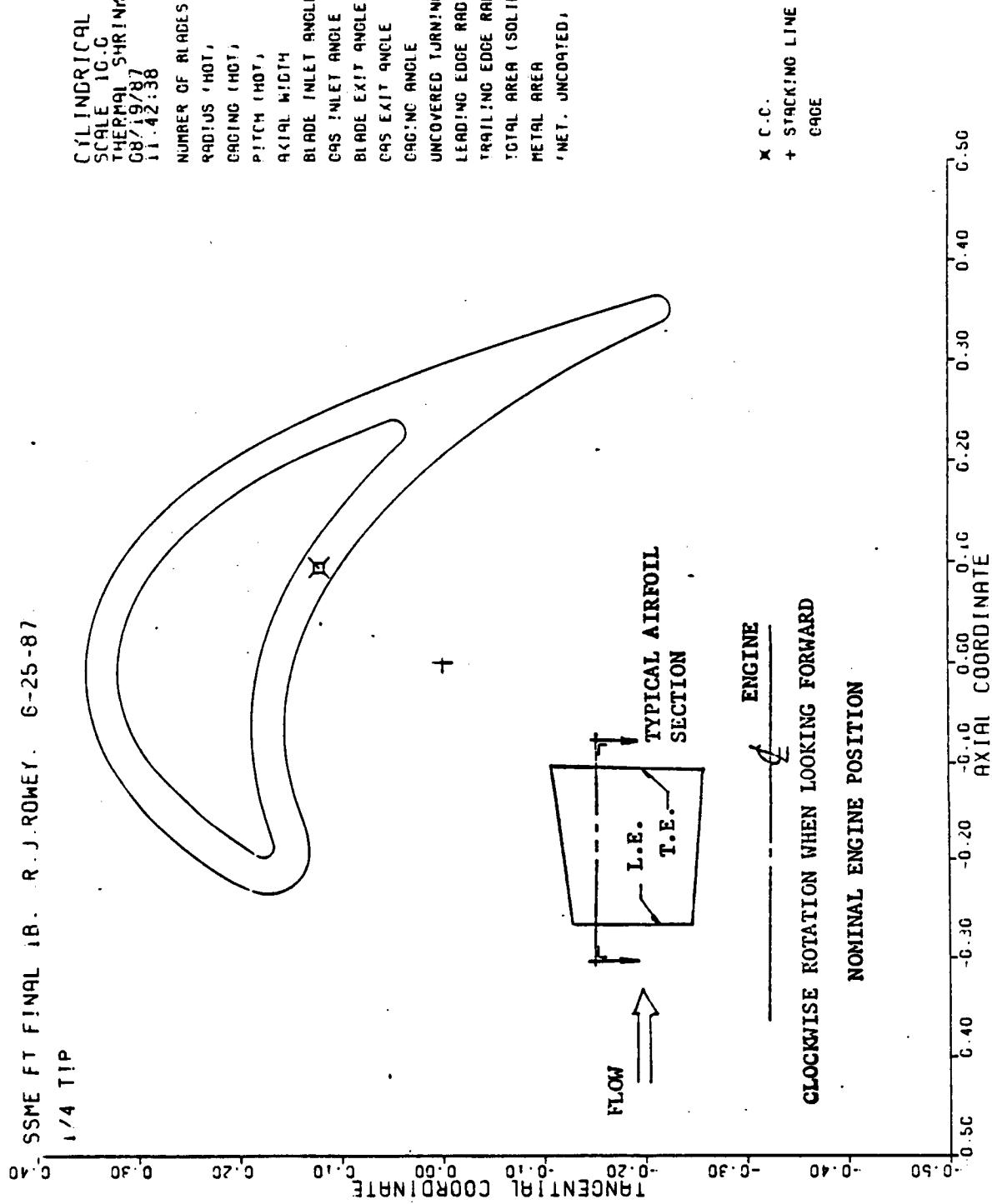
CLOCKWISE ROTATION WHEN LOOKING FORWARD

NOMINAL ENGINE POSITION

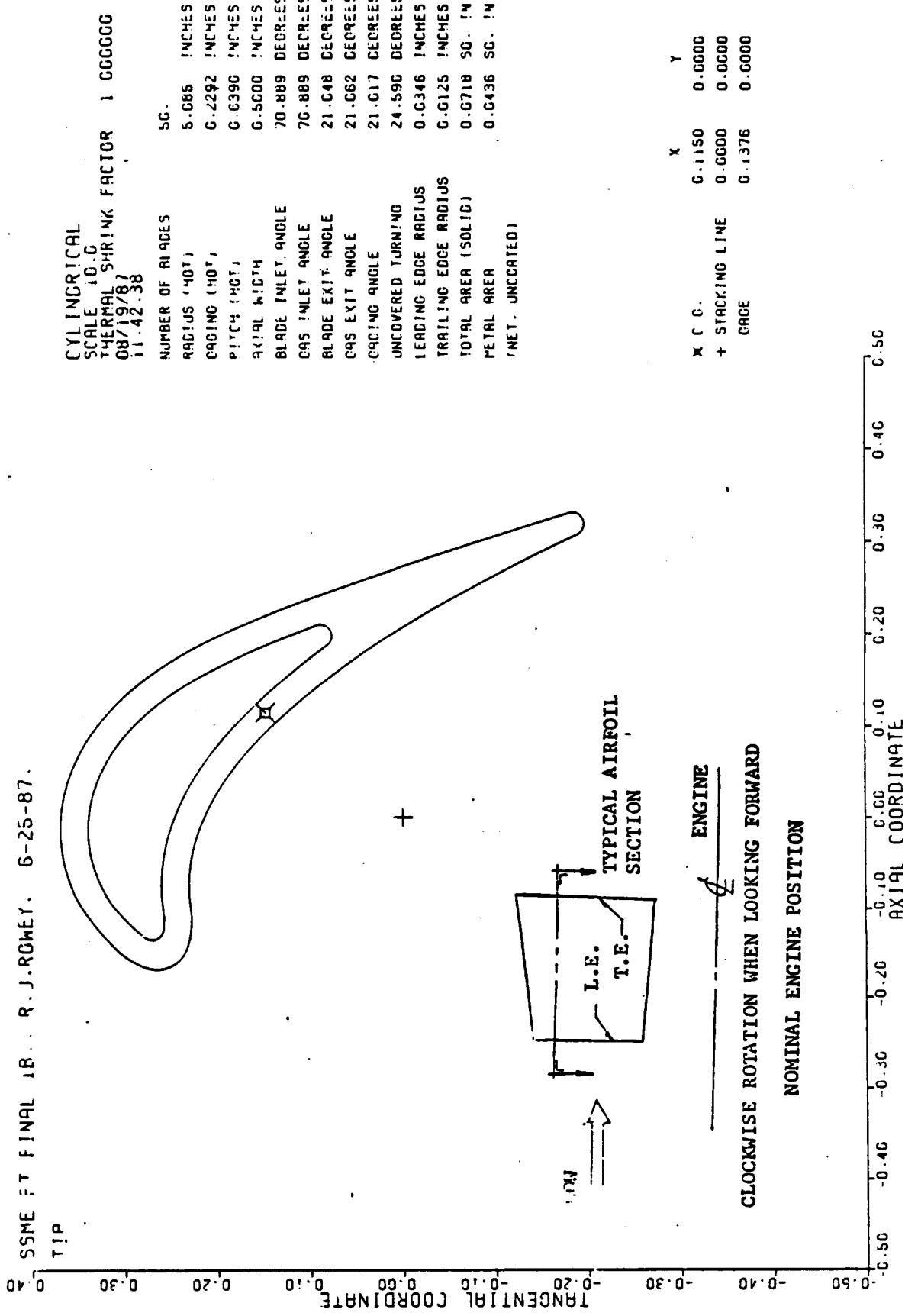


SSME FT FINAL 1B. R. J. ROWLEY. 6-25-87

1/4 TIP



SSME FT FINAL IR. R. J. RGWEI. 6-25-87.



EXTERNAL CONTOUR TITLE - GOME FT FINAL 1B...R.J.ROMEY..10-06-87..
 TD 0 TD REV. 0 PART NO. EMD NO. DATE 06/22/89 TIME 15:27:51
 SUBTITLE HOT RADIUS = 4.12000 COLD RADIUS = 4.10981 THERMAL SHRINK FACTOR = 1.00000 CYLINDRICAL

PRETWIST NOT USED FOR TD PRINTOUT.

PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)
0.0	-0.45053	0.05513	0.01533	-0.45063	-0.00504	0.01522
0.010	-0.45163	0.04414	-0.45163	-0.01540	-0.01181	
0.020	-0.44263	0.05988	-0.44263	-0.02296	-0.02088	
0.030	-0.43363	0.07236	-0.43363	-0.02949	-0.02819	
0.040	-0.42463	0.08431	-0.42463	-0.03621	-0.03411	
0.050	-0.41563	0.09559	-0.41563	-0.04208	-0.04018	
0.060	-0.40663	0.10555	-0.40663	-0.04800	-0.04820	
0.070	-0.39763	0.11510	-0.39763	-0.05265		
0.080	-0.38863	0.12418	-0.38863	-0.05804		
0.090	-0.37963	0.13109	-0.37963	-0.06264		
0.100	-0.37063	0.13955	-0.37063	-0.06351		
0.125	-0.34813	0.15927	-0.34813	-0.01591		
0.150	-0.31563	0.17778	-0.31563	-0.06898		
0.175	-0.30313	0.19313	-0.30313	-0.00290		
0.200	-0.28063	0.20754	-0.28063	0.00238		
0.225	-0.25813	0.22040	-0.25813	0.00683		
0.250	-0.23563	0.23173	-0.23563	0.01064		
0.275	-0.21313	0.24178	-0.21313	0.01366		
0.300	-0.19063	0.25027	-0.19063	0.01547		
0.325	-0.16813	0.25766	-0.16813	0.01757		
0.350	-0.14563	0.26506	-0.14563	0.01847		
0.375	-0.12313	0.26826	-0.12313	0.01868		
0.400	-0.10063	0.27178	-0.10063	0.01819		
0.425	-0.07813	0.27745	-0.07813	0.01701		
0.450	-0.05563	0.27485	-0.05563	0.01513		
0.475	-0.03313	0.27444	-0.03313	0.01254		
0.500	-0.01063	0.27278	-0.01063	0.00423		
0.525	0.01187	0.26976	0.01187	0.00518		
0.550	0.03437	0.26558	0.03437	0.00053		
0.575	0.05687	0.25958	0.05687	-0.00523		
0.600	0.07937	0.25229	0.07937	-0.01165		
0.625	0.10187	0.24365	0.10187	-0.01845		
0.650	0.12437	0.23175	0.12437	-0.02711		
0.675	0.14687	0.22008	0.14687	-0.03625		
0.700	0.16937	0.20668	0.16937	-0.04643		
0.725	0.19187	0.19017	0.19187	-0.05773		
0.750	0.21437	0.17149	0.21437	-0.07028		
0.775	0.23687	0.15012	0.23687	-0.08415		
0.800	0.25937	0.12563	0.25937	-0.09958		
0.825	0.28187	0.09738	0.28187	-0.11679		
0.850	0.30437	0.08448	0.30437	-0.11609		
0.875	0.32687	0.06249	0.32687	-0.15794		
0.900	0.34937	-0.01775	0.34937	-0.18305		
0.910	0.35837	-0.03730	0.35837	-0.19424		
0.925	0.36737	-0.05776	0.36737	-0.20823		
0.930	0.37637	-0.07968	0.37637	-0.21917		
0.940	0.38537	-0.10253	0.38537	-0.23325		
0.950	0.39437	-0.12647	0.39437	-0.24874		
0.960	0.40337	-0.15149	0.40337	-0.26611		
0.970	0.41237	-0.17755	0.41237	-0.28600	-0.27892	
0.980	0.42137	-0.20463	0.42137	-0.30979	-0.28172	
0.990	0.43037	-0.23267	0.43037	-0.34082	-0.27952	
1.000	0.43937	-0.26164	0.43937	-0.37978	-0.26446	

NO. 1 CORE CONTOUR TITLE - GOME FT FINAL 1B...R.J.ROMEY..10-06-87..
 TD 0 TD REV. 0 PART NO. EMD NO. DATE 06/22/89 TIME 15:27:51
 SUBTITLE HOT RADIUS = 4.12000 COLD RADIUS = 4.10981 THERMAL SHRINK FACTOR = 1.00000 CYLINDRICAL

PRETWIST NOT USED FOR TD PRINTOUT.

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0.020	-0.36414	0.06893	-0.36414	0.03799	0.03826	
0.030	-0.35744	0.07535	-0.35744	0.03987		
0.040	-0.35075	0.08177	-0.35075	0.04176		
0.050	-0.34405	0.08817	-0.34405	0.04362		
0.060	-0.33735	0.09450	-0.33735	0.04549		
0.070	-0.33066	0.10071	-0.33066	0.04734		
0.080	-0.32397	0.10614	-0.32397	0.04914		
0.090	-0.31727	0.11255	-0.31727	0.05087		
0.100	-0.31058	0.11817	-0.31058	0.05253		
0.125	-0.29384	0.13162	-0.29384	0.05641		
0.150	-0.27711	0.14409	-0.27711	0.05938		
0.175	-0.26037	0.15448	-0.26037	0.06296		
0.200	-0.24363	0.16483	-0.24363	0.06566		
0.225	-0.22689	0.17404	-0.22689	0.06799		
0.250	-0.21101	0.18133	-0.21101	0.06943		
0.275	-0.19342	0.18974	-0.19342	0.07153		
0.300	-0.17668	0.19627	-0.17668	0.07276		
0.325	-0.15994	0.20197	-0.15994	0.07363		
0.350	-0.14321	0.20626	-0.14321	0.07414		
0.375	-0.12667	0.21103	-0.12667	0.07429		
0.400	-0.10973	0.21420	-0.10973	0.07409		
0.425	-0.09299	0.21665	-0.09299	0.07353		
0.450	-0.07628	0.21800	-0.07628	0.07261		
0.475	-0.05952	0.21910	-0.05952	0.07133		
0.500	-0.04278	0.21910	-0.04278	0.06968		
0.525	-0.02604	0.21820	-0.02604	0.06768		
0.550	-0.00921	0.21681	-0.00921	0.06521		
0.575	0.00743	0.21406	0.00743	0.06258		
0.600	0.02417	0.21062	0.02417	0.05942		
0.625	0.04091	0.20627	0.04091	0.05589		
0.650	0.05768	0.20185	0.05768	0.05178		
0.675	0.07438	0.19463	0.07438	0.04763		
0.700	0.09112	0.18717	0.09112	0.04288		
0.725	0.10784	0.17880	0.10784	0.03768		
0.750	0.12455	0.17019	0.12455	0.03294		
0.775	0.14133	0.15824	0.14133	0.02573		
0.800	0.15807	0.14546	0.15807	0.01433		
0.825	0.17481	0.13218	0.17481	0.01222		
0.850	0.19159	0.11611	0.19159	0.00525		
0.875	0.20828	0.09459	0.20828	-0.00361		
0.900	0.22502	0.07993	0.22502	-0.01245		
0.910	0.23171	0.07148	0.23171	-0.01613		
0.920	0.23941	0.06299	0.23941	-0.01911		
0.930	0.24510	0.05323	0.24510	-0.02384		
0.940	0.25180	0.04338	0.25180	-0.02782		
0.950	0.25849	0.03300	0.25849	-0.03183		
0.960	0.26517	0.02300	0.26517	-0.03581		
0.970	0.27188	0.01045	0.27188	-0.03985		
0.980	0.27858	-0.00132	0.27858	-0.04383	-0.04228	
0.990	0.28527	-0.01381	0.28527	-0.04782	-0.04088	
1.000	0.29197	-0.02084	0.29197	-0.05182	-0.03830	

EXTERNAL CONTOUR TITLE - SGME FT FINAL 1B...R.J.ROMEY..10-06-87...
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PRETWIST NOT USED FOR TD PRINTOUT.

PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)			
0.0	-0.28783	0.05502	0.04018	-0.38783	0.02455	0.0-0.019			
0.010	-0.27983	0.07444		-0.37983	0.01362	0.02001			
0.020	-0.37183	0.09067		-0.37183	0.00632	0.01045			
0.030	-0.36383	0.10289		-0.36383	0.00275	0.00572			
0.040	-0.25583	0.11571		-0.35583	0.00621	0.00183			
0.050	-0.24783	0.12726		-0.34783	0.00068	-0.00009			
0.060	-0.23983	0.13855		-0.33983	0.00062	-0.00057			
0.070	-0.22783	0.14884		-0.32783	0.00000		L.E. CIRCLE (X-Y-R)	-0.36108	0.06818
0.080	-0.21983	0.15614		-0.31983	0.00171		L.E. CIRCLE (X-Y-R)	0.38942	-0.24600
0.090	-0.21583	0.16896		-0.31583	0.00378		L.E. TOP TANG. PT. (X-Y)	-0.38436	0.06383
0.100	-0.20783	0.17833		-0.30783	0.00638		L.E. BOTTOM TANG. PT. (X-Y)	-0.37556	-0.00306
0.125	-0.28783	0.20007		-0.28783	0.01688		L.E. TOP TANG. PT. (X-Y)	0.41146	-0.11115
0.150	-0.26183	0.21955		-0.26183	0.02364		T.E. BOTTOM TANG. PT. (X-Y)	0.38122	-0.25293
0.175	-0.24783	0.23704		-0.24783	0.03159		NOSE POINT (X-Y)	-0.37858	0.01825
0.200	-0.22783	0.25258		-0.22783	0.03854		TAIL POINT (X-Y)	0.40176	-0.26119
0.225	-0.20783	0.26621		-0.20783	0.04461		NET CROSS-SECT. AREA (EXCL. COATING)	0.08747	
0.250	-0.18783	0.2781		-0.18783	0.05176		GAGING (X-Y-LAMEA)	0.21818	0.17031
0.275	-0.16783	0.28867		-0.16783	0.05402		GAGING ANGLE (DEG.)	21.062	0.17030
0.300	-0.14783	0.29744		-0.14783	0.05740		CTR. OF GRAV. (INCL. COATING) (X-Y)	0.02861	0.08399
0.325	-0.12783	0.30468		-0.12783	0.06043		RAD. KEP. PT. (X-Y)	0.0	0.0
0.350	-0.10783	0.31057		-0.10783	0.06157		INLET ANGLE (DEG.)	53.331	
0.375	-0.08783	0.31454		-0.08783	0.06240		INLET WEDGE ANGLE (DEG.)	62.339	
0.400	-0.06783	0.31735		-0.06783	0.06239		EXIT ANGLE (DEG.)	21.126	
0.425	-0.04783	0.31845		-0.04783	0.06146		EXIT WEDGE ANGLE (DEG.)	8.331	
0.450	-0.02783	0.31884		-0.02783	0.05664		UNCOVERED TURNING ANGLE (DEG.)	21.152	
0.475	-0.00783	0.31885		-0.00783	0.05704		AXIAL CHORD	0.80000	
0.500	0.01217	0.31372		0.01217	0.05250		ACTUAL CHORD	0.85580	
0.525	0.03217	0.30903		0.03217	0.04903		PITCH	0.5-305	
0.550	0.05217	0.30517		0.05217	0.04663		NO. OF FOILS	50	
0.575	0.07217	0.29515		0.07217	0.04372				
0.600	0.09217	0.28575		0.09217	0.04297				
0.625	0.11217	0.27466		0.11217	0.04214				
0.650	0.13217	0.26160		0.13217	0.04116				
0.675	0.15217	0.24708		0.15217	0.04018				
0.700	0.17217	0.23303		0.17217	0.04000				
0.725	0.19217	0.21132		0.19217	0.03901				
0.750	0.21217	0.18478		0.21217	0.03881				
0.775	0.23217	0.16529		0.23217	0.03509				
0.800	0.25217	0.13744		0.25217	0.03209				
0.825	0.27217	0.10570		0.27217	0.03001				
0.850	0.29217	0.06717		0.29217	0.01150				
0.875	0.31217	0.02948		0.31217	-0.13467				
0.900	0.33217	-0.01577		0.33217	-0.16723				
0.910	0.34017	-0.03524		0.34017	-0.17022				
0.930	0.35617	-0.05558		0.35617	-0.19186				
0.950	0.35617	-0.07664		0.35617	-0.20522				
0.960	0.36417	-0.09826		0.36417	-0.21958				
0.970	0.38817	-0.16793		0.38817	-0.24854				
0.980	0.39817	-0.19552		0.39817	-0.28792				
0.990	0.40417	-0.21775		0.40417	-0.31061				
1.000	0.41217	-0.24457		0.41217	-0.34600				

NO. 1 CORE CONTOUR TITLE - SGME FT FINAL 1B...R.J.ROMEY..10-06-87...
 TD 0 TD REV. 0 PART NO. EMD NO. DATE 06/22/89 TIME 15:27:51 CYLINDRICAL
 SUBTITLE HOT RADIUS = 4.36125 COLD RADIUS = 4.36100 THERMAL SHRINK FACTOR = 1.00000

PRETWIST NOT USED FOR TD PRINTOUT.

PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)			
0.0	-0.22560	0.06880	0.06210	-0.32560	0.06879	0.0-0.010			
0.010	-0.32256	0.07444		-0.32256	0.06936	0.05341			
0.020	-0.31653	0.08254		-0.31653	0.05190	0.05238			
0.030	-0.31050	0.09468		-0.31050	0.05448				
0.040	-0.29646	0.09844		-0.29646	0.05705				
0.050	-0.28843	0.10638		-0.28843	0.05959				
0.060	-0.28240	0.11418		-0.28240	0.06210				
0.070	-0.28637	0.12179		-0.28637	0.06453		L.E. CIRCLE (X-Y-R)	-0.31780	0.0-0.01080
0.080	-0.29303	0.12777		-0.29303	0.06687		L.E. CIRCLE (X-Y-R)	0.36240	-0.08555
0.090	-0.27430	0.13620		-0.27430	0.06911		L.E. TOP TANG. PT. (X-Y)	-0.32635	0.06970
0.100	-0.26827	0.14319		-0.26827	0.07125		L.E. BOTTOM TANG. PT. (X-Y)	-0.31256	0.05317
0.125	-0.25318	0.15988		-0.25318	0.07625		L.E. TOP TANG. PT. (X-Y)	0.37570	-0.08051
0.150	-0.22910	0.17758		-0.22910	0.08076		L.E. BOTTOM TANG. PT. (X-Y)	0.25502	-0.01519
0.175	-0.22302	0.19164		-0.22302	0.08474		NOSE POINT (X-Y)	-0.32634	0.05650
0.200	-0.20794	0.20568		-0.20794	0.08816		TAIL POINT (X-Y)	0.37000	-0.01502
0.225	-0.19286	0.21795		-0.19286	0.09121				
0.250	-0.17777	0.23001		-0.17777	0.09410				
0.275	-0.16269	0.23903		-0.16269	0.09683				
0.300	-0.14761	0.24775		-0.14761	0.09913				
0.325	-0.13253	0.25506		-0.13253	0.10090				
0.350	-0.11743	0.26111		-0.11743	0.10208				
0.375	-0.10236	0.26647		-0.10236	0.10308				
0.400	-0.08726	0.27066		-0.08726	0.10350				
0.425	-0.07220	0.27357		-0.07220	0.10348				
0.450	-0.05712	0.27544		-0.05712	0.10380				
0.475	-0.04203	0.27584		-0.04203	0.10208				
0.500	-0.02695	0.27554		-0.02695	0.10070				
0.525	-0.01187	0.27428		-0.01187	0.09887				
0.550	0.00521	0.27118		0.00521	0.09656				
0.575	0.01830	0.26867		0.01830	0.09378				
0.600	0.03338	0.26431		0.03338	0.09053				
0.625	0.04846	0.25842		0.04846	0.08681				
0.650	0.06354	0.25146		0.06354	0.08233				
0.675	0.07862	0.24484		0.07862	0.07784				
0.700	0.09371	0.23112		0.09371	0.07250				
0.725	0.10874	0.21624		0.10874	0.06684				
0.750	0.12387	0.19708		0.12387	0.06050				
0.775	0.13895	0.20256		0.13895	0.05352				
0.800	0.15404	0.18863		0.15404	0.04614				
0.825	0.16912	0.17341		0.16912	0.03855				
0.850	0.18420	0.15741		0.18420	0.03183				
0.875	0.19928	0.13804		0.19928	0.02222				
0.900	0.21430	0.11619		0.21430	0.01310				
0.910	0.22040	0.10643		0.22040	0.00923				
0.925	0.22643	0.09711		0.22643	0.00933				
0.930	0.23246	0.08531		0.23246	0.00912				
0.940	0.23850	0.07406		0.23850	-0.00310				
0.950	0.24453	0.06232		0.24453	-0.00743				
0.960	0.25050	0.05010		0.25050	-0.01127				
0.970	0.25659	0.03755		0.25659	-0.01637	-0.01620			
0.980	0.26263	0.02452		0.26263	-0.02091	-0.01745			
0.990	0.26866	0.01106		0.26866	-0.02545	-0.01546			
1.000	0.27468	0.00235	0.00235	0.27468	-0.03000	-0.01501			

EXTERNAL CONTOUR TD 0 TD REV. 0 PART NO. SUBTITLE		TITLE - GOME FT FINAL 1B...R.J.RONEY..10-06-87... EMD NO. DATE 06/22/89 TIME 15:27:51 HOT RADIUS = 4.60250 COLD RADIUS = 4.58617 THERMAL SHRINK FACTOR = 1.00000 CYLINDRICAL				
PRETWIST NOT USED FOR TD PRINTOUT.						
PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)
0.0	-0.31784	0.11288	0.10481	-0.31784	0.08421	0.10481
0.010	-0.31094	0.13084		-0.31094	0.07414	0.08077
0.020	-0.30384	0.14588		-0.30384	0.06714	0.07228
0.030	-0.29684	0.15800		-0.29684	0.06283	0.06868
0.040	-0.28984	0.1655		-0.28984	0.06100	0.06328
0.050	-0.28284	0.1801		-0.28284	0.05492	0.06110
0.060	-0.27584	0.19055		-0.27584	0.05070	0.06010
0.070	-0.26884	0.20065		-0.26884	0.06019	0.06021 L.E. CIRCLE (X-Y-R)
0.080	-0.26184	0.20484		-0.26184	0.05127	T.E. CIRCLE (X-Y-R) -0.27204 0.10481 0.06480
0.090	-0.25484	0.21897		-0.25484	0.05287	0.36716 -0.22217 0.01560
0.100	-0.24784	0.22763		-0.24784	0.06496	L.E. TOP TANG. PT. (X-Y) -0.31483 0.12095
0.125	-0.22074	0.24762		-0.22074	0.07197	L.E. BOTTOM TANG. PT. (X-Y) -0.27701 0.08042
0.150	-0.19534	0.2554		-0.19534	0.08030	T.E. TOP TANG. PT. (X-Y) 0.26150 -0.21508
0.175	-0.16534	0.28125		-0.16534	0.08767	T.E. BOTTOM TANG. PT. (X-Y) 0.35358 -0.22964
0.200	-0.17784	0.29518		-0.17784	0.09401	
0.225	-0.16074	0.30724		-0.16074	0.09935	NOSE POINT (X-Y) -0.30819 0.07704
0.250	-0.14154	0.31769		-0.14154	0.10276	TAIL POINT (X-Y) 0.37155 -0.23517
0.275	-0.12534	0.32669		-0.12534	0.10716	
0.300	-0.10784	0.33400		-0.10784	0.10963	NET CROSS-SEC'T. AREA (EXCL. COATING) 0.06846
0.325	-0.09034	0.33974		-0.09034	0.11116	
0.350	-0.07184	0.34410		-0.07184	0.11175	GAGING (A-Y-LAMEDET)
0.375	-0.05534	0.34834		-0.05534	0.11141	GAGING ANGLE (DEG.) 0.18172 0.21496 0.16646
0.400	-0.03784	0.34834		-0.03784	0.11016	
0.425	-0.02034	0.34830		-0.02034	0.10793	CTR. OF GRAV. (INCL. COATING) (X-Y) 0.05446 0.11040
0.450	-0.00184	0.34680		-0.00184	0.10478	KAD. REF. PT. (X-Y) 0.0 0.0
0.475	0.01466	0.34385		0.01466	0.10066	INLET ANGLE (DEG.) 51.9%
0.500	0.03214	0.33940		0.03214	0.09558	INLET WEDGE ANGLE (DEG.) 61.158
0.525	0.06264	0.33344		0.06264	0.08951	EXIT ANGLE (DEG.) 21.0%
0.550	0.08716	0.3354		0.08716	0.08243	EXIT WEDGE ANGLE (DEG.) 6.0%
0.575	0.08466	0.31679		0.08466	0.07432	UNCOVERED TURNING ANGLE (DEG.) 20.61%
0.600	0.10216	0.20586		0.10216	0.05616	AXIAL CHORD 0.70000
0.625	0.11466	0.19337		0.11466	0.05665	ACTUAL CHORD 0.77917
0.650	0.12716	0.17888		0.12716	0.05345	PITCH 0.5787
0.675	0.15466	0.1674		0.15466	0.05080	NO. OF FOILS 50
0.700	0.17216	0.24370		0.17216	0.04642	
0.725	0.18866	0.22261		0.18866	0.04017	
0.750	0.20716	0.19834		0.20716	0.03485	
0.775	0.22466	0.17210		0.22466	0.03300	
0.800	0.24216	0.14210		0.24216	0.03274	
0.825	0.25966	0.10863		0.25966	0.03742	
0.850	0.27716	0.07157		0.27716	0.03976	
0.875	0.29466	0.02088		0.29466	0.12326	
0.900	0.31216	-0.01224		0.31216	0.15129	
0.910	0.31916	-0.03198		0.31916	0.16325	
0.920	0.33616	-0.05111		0.33616	0.1758	
0.930	0.33716	-0.07081		0.33716	0.18864	
0.940	0.34016	-0.08096		0.34016	0.20212	
0.950	0.34716	-0.11159		0.34716	0.21616	
0.960	0.35516	-0.13388		0.35516	0.22075	
0.970	0.36116	-0.15416		0.36116	0.24826	
0.980	0.35816	-0.17806		0.35816	0.26237	
0.990	0.35751	-0.19836		0.35751	0.27554	
1.000	0.35816	-0.21102		0.35816	0.28711	
NO. 1 CORE CONTOUR TD 0 TD REV. 0 PART NO. SUBTITLE						
TITLE - GOME FT FINAL 1B...R.J.RONEY..10-06-87... EMD NO. DATE 06/22/89 TIME 15:27:51 HOT RADIUS = 4.60250 COLD RADIUS = 4.58617 THERMAL SHRINK FACTOR = 1.00000 CYLINDRICAL						
PRETWIST NOT USED FOR TD PRINTOUT.						
PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)
0.0	-0.31774	0.11475	0.11163	-0.31774	0.09625	0.11163
0.010	-0.29848	0.15248		-0.29848	0.09855	0.10280
0.020	-0.28522	0.13025		-0.28522	0.10086	0.10160
0.030	-0.26796	0.13801		-0.26796	0.10318	
0.040	-0.25570	0.14578		-0.25570	0.10550	
0.050	-0.24744	0.15367		-0.24744	0.10780	
0.060	-0.24216	0.16102		-0.24216	0.11006	
0.070	-0.23693	0.16836		-0.23693	0.11228	L.E. CIRCLE (X-Y-R) -0.26364 0.11163 0.01005
0.080	-0.22167	0.17753		-0.22167	0.11464	0.26765 0.08912 0.01257
0.090	-0.22641	0.18228		-0.22641	0.11649	
0.100	-0.22115	0.18887		-0.22115	0.11849	L.E. TOP TANG. PT. (X-Y) -0.27199 0.11731
0.125	-0.20800	0.20465		-0.20800	0.13118	L.E. BOTTOM TANG. PT. (X-Y) -0.25943 0.10244
0.150	-0.17688	0.2046		-0.17688	0.13484	T.E. TOP TANG. PT. (X-Y) 0.23148 0.01183
0.175	-0.18171	0.23592		-0.18171	0.13101	T.E. BOTTOM TANG. PT. (X-Y) 0.23122 0.01083
0.200	-0.16856	0.24961		-0.16856	0.13406	
0.225	-0.15541	0.26140		-0.15541	0.13669	NOSE POINT (X-Y) -0.27162 0.10521
0.250	-0.14211	0.27776		-0.14211	0.13925	TAIL POINT (X-Y) 0.24677 0.01582
0.275	-0.12912	0.28155		-0.12912	0.14166	
0.300	-0.11597	0.28978		-0.11597	0.14363	
0.325	-0.10282	0.29653		-0.10282	0.14503	
0.350	-0.08768	0.30610		-0.08768	0.14654	
0.375	-0.07653	0.30679		-0.07653	0.14664	
0.400	-0.06338	0.31043		-0.06338	0.14639	
0.425	-0.05023	0.31273		-0.05023	0.14590	
0.450	-0.03700	0.31376		-0.03700	0.14550	
0.475	-0.02394	0.31359		-0.02394	0.14348	
0.500	-0.01079	0.31277		-0.01079	0.14155	
0.525	0.00236	0.31091		0.00236	0.13916	
0.550	0.01350	0.29800		0.01350	0.12823	
0.575	0.0285	0.29413		0.0285	0.12282	
0.600	0.04180	0.29922		0.04180	0.12892	
0.625	0.05495	0.29529		0.05495	0.12452	
0.650	0.06902	0.2867		0.06902	0.11700	
0.675	0.08124	0.27819		0.08124	0.11414	
0.700	0.09439	0.26898		0.09439	0.10816	
0.725	0.10754	0.25863		0.10754	0.10165	
0.750	0.12308	0.24702		0.12308	0.09455	
0.775	0.13382	0.23402		0.13382	0.08677	
0.800	0.14696	0.21963		0.14696	0.07863	
0.825	0.16012	0.20400		0.16012	0.07037	
0.850	0.18642	0.18813		0.18642	0.05289	
0.875	0.19857	0.16589		0.19857	0.04224	
0.900	0.20483	0.15597		0.20483	0.03916	
0.920	0.21004	0.15562		0.21004	0.03498	
0.930	0.21535	0.14642		0.21535	0.03064	
0.940	0.22061	0.10333		0.22061	0.02820	
0.950	0.22586	0.09166		0.22586	0.02151	
0.960	0.22912	0.07746		0.22912	0.01403	
0.970	0.23238	0.07331		0.23238	0.01212	0.01402
0.980	0.24164	0.05467		0.24164	0.00721	0.01375
0.990	0.24690	0.04173		0.24690	0.00245	0.01592
1.000	0.25610	0.03590	0.02612	0.25610	0.00140	0.01611

EXTERNAL CONTOUR TITLE - SSMF FT FINAL 1B...R.J.ROWLEY..10-06-87..
 TD 0 TD REV. 0 PART NO. EMD NO. DATE 06/22/89 TIME 15:27:51
 SUBTITLE HOT RADIUS = 4.84375 COLD RADIUS = 4.82433 THERMAL SHRINK FACTOR = 1.00000 CYLINDRICAL

PRETWIST NOT USED FOR TD PRINTOUT.

PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)				
0.0	-0.25058	0.19575	0.18811	-0.25058	0.171e1	0.18811				
0.010	-0.24438	0.21058		-0.24438	0.162e1	0.1ee81				
0.020	-0.23838	0.22285		-0.23838	0.15e1	0.15e21				
0.030	-0.23238	0.2286		-0.23238	0.15277	0.15427				
0.040	-0.22638	0.24186		-0.22638	0.14887	0.15062				
0.050	-0.22038	0.2505		-0.22038	0.14834	0.14870				
0.060	-0.21428	0.25831		-0.21428	0.14752	0.14758				
0.070	-0.20838	0.26584		-0.20838	0.14778					
0.080	-0.20238	0.27301		-0.20238	0.14756		L.E. CIRCLE (X.Y.R)	-0.20296	0.18811	0.01082
0.090	-0.19638	0.27983		-0.19638	0.14616		L.E. CIRCLE (X.Y.R)	0.13588	-0.14651	0.01375
0.100	-0.19038	0.28623		-0.19038	0.14615		L.E. TOP TANG. PT. (X.Y)	-0.24739	0.20344	
0.125	-0.17538	0.29121		-0.17538	0.15215		L.E. BOTTOM TANG. PT. (X.Y)	-0.31094	0.14722	
0.150	-0.16238	0.29423		-0.16238	0.15646		T.E. TOP TANG. PT. (X.Y)	0.34902	-0.14248	
0.175	-0.14538	0.29572		-0.14538	0.16108		T.E. BOTTOM TANG. PT. (X.Y)	0.32343	-0.20235	
0.200	-0.13038	0.29560		-0.13038	0.16170		NOSE POINT (X.Y)	-0.24177	0.1b588	
0.225	-0.11538	0.29462		-0.11538	0.16274		TAIL POINT (X.Y)	0.14082	-0.10932	
0.250	-0.10038	0.29105		-0.10038	0.16102		NET CROSS-SECT. AREA (EXCL. COATING)	0.05499		
0.275	-0.08538	0.28675		-0.08538	0.16174		GAGING (X.Y.LAPEDA)	0.15463	0.26778	0.01788
0.200	-0.07038	0.28615		-0.07038	0.17151		GAGING ANGLE (DEG.)	20.985		
0.225	-0.05538	0.28424		-0.05538	0.17033		CTR. OF GRAV. (INCL. COATING) (X.Y)	0.07903	0.13740	
0.250	-0.04038	0.28019		-0.04038	0.16821		RAD. REF. PT. (X.Y)	0.0	0.0	
0.275	-0.03438	0.27579		-0.03438	0.16350		INLET ANGLE (DEG.)	58.998		
0.500	-0.04962	0.35143		-0.04962	0.13571		INLET WEDGE ANGLE (DEG.)	69.874		
0.525	-0.04962	0.34441		-0.04962	0.12596		EXIT ANGLE (DEG.)	21.085		
0.550	-0.07962	0.35545		-0.07962	0.11724		EXIT WEDGE ANGLE (DEG.)	8.058		
0.575	-0.09462	0.35596		-0.09462	0.10653		UNCOVERED TURNING ANGLE (DEG.)	21.497		
0.600	-0.10962	0.34432		-0.10962	0.10482		AXIAL CHORD	0.60000		
0.625	-0.12462	0.30088		-0.12462	0.09820		ACTUAL CHORD	0.72197		
0.650	-0.13962	0.28543		-0.13962	0.09820		FILTH	0.00808		
0.675	-0.15462	0.26774		-0.15462	0.09344		NO. OF FOILS	50		
0.700	-0.16962	0.24771		-0.16962	0.08767					
0.725	-0.18462	0.22504		-0.18462	0.08051					
0.750	-0.19962	0.19964		-0.19962	0.07024					
0.775	-0.21462	0.17142		-0.21462	-0.01715					
0.800	-0.22962	0.14035		-0.22962	-0.03788					
0.825	-0.24462	0.10644		-0.24462	-0.05846					
0.850	-0.25962	0.08446		-0.25962	-0.08346					
0.875	-0.27462	0.03087		-0.27462	-0.10852					
0.900	-0.28962	-0.01051		-0.28962	-0.13523					
0.910	-0.29562	-0.02766		-0.29562	-0.16642					
0.920	-0.30162	-0.05513		-0.30162	-0.15740					
0.930	-0.30762	-0.08289		-0.30762	-0.16970					
0.940	-0.31362	-0.08094		-0.31362	-0.18182					
0.950	-0.31962	-0.09926		-0.31962	-0.19427					
0.960	-0.32562	-0.11786		-0.32562	-0.20705					
0.970	-0.33162	-0.13667		-0.33162	-0.22018					
0.980	-0.33762	-0.15571		-0.33762	-0.23358					
0.990	-0.34362	-0.17498		-0.34362	-0.24712					
1.000	-0.34962	-0.19451		-0.34962	-0.25767					

NO. 1 CORE CONTOUR TITLE - SSMF FT FINAL 1B...R.J.ROWLEY..10-06-87..
 TD 0 TD REV. 0 PART NO. EMD NO. DATE 06/22/89 TIME 15:27:51 CYLINDRICAL
 SUBTITLE HOT RADIUS = 4.84375 COLD RADIUS = 4.82433 THERMAL SHRINK FACTOR = 1.00000

PRETWIST NOT USED FOR TD PRINTOUT.

PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)				
0.0	-0.21057	0.19165	0.19167	-0.21057	0.17874	0.17876				
0.010	-0.20950	0.20225		-0.20950	0.17996	0.18409				
0.020	-0.20482	0.20841		-0.20482	0.18117	0.18224				
0.030	-0.20043	0.21452		-0.20043	0.18239	0.18239				
0.040	-0.19605	0.22558		-0.19605	0.18281					
0.050	-0.19166	0.22668		-0.19166	0.18484					
0.060	-0.18728	0.23222		-0.18728	0.18610					
0.070	-0.18284	0.23776		-0.18284	0.18738		L.E. CIRCLE (X.Y.R)	-0.20299	0.19267	0.01059
0.080	-0.17851	0.24307		-0.17851	0.18867		T.E. CIRCLE (X.Y.R)	0.11208	-0.08296	0.01285
0.090	-0.17412	0.24815		-0.17412	0.18996					
0.100	-0.16973	0.25303		-0.16973	0.19125		L.E. TOP TANG. PT. (X.Y)	-0.21163	0.19880	
0.125	-0.15877	0.26457		-0.15877	0.19434		L.E. BOTTOM TANG. PT. (X.Y)	-0.20016	0.18246	
0.150	-0.14781	0.27304		-0.14781	0.19706		L.E. TOP TANG. PT. (X.Y)	0.21605	0.06351	
0.175	-0.13685	0.28675		-0.13685	0.19919		L.E. BOTTOM TANG. PT. (X.Y)	0.20261	0.05527	
0.200	-0.12588	0.29629		-0.12588	0.20082					
0.225	-0.11149	0.30443		-0.11149	0.20208		NOSE POINT (X.Y)	-0.21166	0.18659	
0.250	-0.10076	0.31184		-0.10076	0.20319		TAIL POINT (X.Y)	0.11887	0.05005	
0.275	-0.09500	0.31801		-0.09500	0.20397					
0.300	-0.08820	0.32336		-0.08820	0.20436					
0.325	-0.07107	0.32760		-0.07107	0.20426					
0.350	-0.06071	0.33071		-0.06071	0.20572					
0.375	-0.05614	0.33350		-0.05614	0.20675					
0.400	-0.052818	0.33526		-0.052818	0.20133					
0.425	-0.05272	0.33549		-0.05272	0.19847					
0.450	-0.04918	0.33748		-0.04918	0.19715					
0.475	-0.04529	0.33465		-0.04529	0.19640					
0.500	-0.05067	0.33282		-0.05067	0.19120					
0.525	-0.04683	0.33201		-0.04683	0.18755					
0.550	-0.04734	0.32876		-0.04734	0.18545					
0.575	-0.03856	0.32247		-0.03856	0.17885					
0.600	-0.04952	0.31732		-0.04952	0.17581					
0.625	-0.06048	0.31129		-0.06048	0.16831					
0.650	-0.07144	0.29933		-0.07144	0.16334					
0.675	-0.08241	0.27644		-0.08241	0.15588					
0.700	-0.09337	0.28754		-0.09337	0.14894					
0.725	-0.10433	0.27757		-0.10433	0.14153					
0.750	-0.11533	0.26642		-0.11533	0.13533					
0.775	-0.12626	0.25398		-0.12626	0.12504					
0.800	-0.13722	0.26018		-0.13722	0.11821					
0.825	-0.14818	0.22508		-0.14818	0.10710					
0.850	-0.17011	0.19041		-0.17011	0.08789					
0.900	-0.18107	0.18971		-0.18107	0.07751					
0.910	-0.18546	0.16067		-0.18546	0.07318					
0.920	-0.19394	0.15724		-0.19394	0.06873					
0.930	-0.19423	0.16146		-0.19423	0.06422					
0.940	-0.19881	0.13137		-0.19881	0.05959					
0.950	-0.20200	0.12101		-0.20200	0.05485	0.05487				
0.960	-0.20738	0.11044		-0.20738	0.05005	0.05006				
0.970	-0.21177	0.09863		-0.21177	0.04513	0.05112				
0.980	-0.21615	0.08868		-0.21615	0.04021	0.05178				
0.990	-0.22054	0.07754		-0.22054	0.03750	0.05429				
1.000	-0.22400	0.06657		-0.22400	0.03357	0.05424				

EXTERNAL CONTOUR TITLE - SOME FT FINAL 1B...R.J.ROWNEY..10-06-87..
 TD 0 TD REV. 0 PART NO. EMD NO. DATE 06/22/89 TIME 15:27:51 CYLINDRICAL
 SUBTITLE HOT RADIUS = 5.08500 COLD RADIUS = 5.08500 THERMAL SHRINK FACTOR = 1.00000

PRETWIST NOT USED FOR TD PRINTOUT.

PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)
0.0	-0.12845	0.28893	0.28100	-0.16395	0.27021	0.28100
0.010	-0.17845	0.14894		-0.17845	0.16261	0.26575
0.020	-0.17745	0.30402		-0.17345	0.25680	0.25732
0.030	-0.16845	0.31014		-0.16845	0.25705	0.25215
0.040	-0.16345	0.31222		-0.16345	0.25024	
0.050	-0.15845	0.32760		-0.15845	0.24822	
0.060	-0.15345	0.33247		-0.15345	0.24667	
0.070	-0.14845	0.33646		-0.14845	0.24553	
0.080	-0.14345	0.33114		-0.14345	0.24471	
0.090	-0.13845	0.34517		-0.13845	0.24419	
0.100	-0.13345	0.34891		-0.13345	0.24392	
0.125	-0.12145	0.35720		-0.12145	0.24416	
0.150	-0.10845	0.36744		-0.10845	0.24548	
0.175	-0.09445	0.37030		-0.09445	0.24645	
0.200	-0.08345	0.37512		-0.08345	0.24648	
0.225	-0.07165	0.37890		-0.07165	0.24555	
0.250	-0.05645	0.38140		-0.05645	0.24577	
0.275	-0.04645	0.38553		-0.04645	0.24103	
0.300	-0.03245	0.38443		-0.03245	0.23777	
0.325	-0.02145	0.38441		-0.02145	0.23778	
0.350	-0.00845	0.38448		-0.00845	0.23777	
0.375	0.00355	0.38163		0.00355	0.22085	
0.400	0.01605	0.37885		0.01605	0.21350	
0.425	0.02855	0.37912		0.02855	0.20525	
0.450	0.04105	0.37941		0.04105	0.19608	
0.475	0.05255	0.38467		0.05255	0.18601	
0.500	0.06605	0.35784		0.06605	0.17504	
0.525	0.07855	0.34986		0.07855	0.16317	
0.550	0.09105	0.34062		0.09105	0.15050	
0.575	0.10255	0.32999		0.10255	0.13674	
0.600	0.11605	0.31780		0.11605	0.12220	
0.625	0.12855	0.30270		0.12855	0.10678	
0.650	0.14105	0.28769		0.14105	0.09049	
0.675	0.15255	0.26410		0.15255	0.07332	
0.700	0.16605	0.24786		0.16605	0.05520	
0.725	0.17855	0.22582		0.17855	0.03642	
0.750	0.19105	0.19703		0.19105	0.01689	
0.775	0.20355	0.18765		0.20355	-0.00388	
0.800	0.21505	0.17596		0.21505	-0.02529	
0.825	0.22855	0.16122		0.22855	-0.04753	
0.850	0.24105	0.14697		0.24105	-0.07358	
0.875	0.25355	0.13028		0.25355	-0.09446	
0.900	0.26605	0.11752		0.26605	-0.11413	
0.910	0.27105	0.10240		0.27105	-0.12433	
0.930	0.27605	0.08341		0.27605	-0.13464	
0.930	0.28105	0.05405		0.28105	-0.14479	
0.940	0.28605	-0.06980		0.28605	-0.16027	
0.950	0.29105	-0.08565		0.29105	-0.17087	
0.960	0.29605	-0.10160		0.29605	-0.18159	-0.17809
0.970	0.30105	-0.11764		0.30105	-0.19245	-0.18034
0.980	0.30605	-0.13276		0.30605	-0.20342	-0.18034
0.990	0.31105	-0.14495		0.31105	-0.21451	-0.17809
1.000	0.31605	-0.16822		0.31605	-0.21605	-0.16809

NO. 1 CORE CONTOUR TITLE - SOME FT FINAL 1B...R.J.ROWNEY..10-06-87..
 TD 0 TD REV. 0 PART NO. EMD NO. DATE 06/22/89 TIME 15:27:51 CYLINDRICAL
 SUBTITLE HOT RADIUS = 5.08500 COLD RADIUS = 5.08333 THERMAL SHRINK FACTOR = 1.00000

PRETWIST NOT USED FOR TD PRINTOUT.

PCT X	X TOP	Y TOP	(CIRCLE)	X BOT	Y BOT	(CIRCLE)
0.0	-0.15125	0.2107	0.28651	-0.15125	0.27584	0.28651
0.010	-0.14778	0.29517		-0.14778	0.27463	0.27788
0.020	-0.14431	0.29014		-0.14431	0.27440	0.27534
0.030	-0.14084	0.30298		-0.14084	0.27416	0.27422
0.040	-0.13727	0.30684		-0.13727	0.27359	
0.050	-0.13390	0.31014		-0.13390	0.27376	
0.060	-0.13043	0.31244		-0.13043	0.27369	
0.070	-0.12686	0.31655		-0.12686	0.27373	
0.080	-0.12349	0.31467		-0.12349	0.27370	
0.090	-0.12002	0.32021		-0.12002	0.27461	
0.100	-0.11555	0.32479		-0.11555	0.27651	
0.125	-0.10787	0.33064		-0.10787	0.27564	
0.150	-0.09453	0.34278		-0.09453	0.27413	
0.175	-0.09053	0.34027		-0.09053	0.27460	
0.200	-0.08185	0.34410		-0.08185	0.27622	
0.225	-0.07218	0.34730		-0.07218	0.27568	
0.250	-0.06430	0.35197		-0.06430	0.27478	
0.275	-0.05583	0.35197		-0.05583	0.27347	
0.300	-0.04715	0.35346		-0.04715	0.27178	
0.325	-0.03848	0.35438		-0.03848	0.26970	
0.350	-0.03030	0.35476		-0.03030	0.26738	
0.375	-0.02113	0.35661		-0.02113	0.26444	
0.400	-0.01245	0.35593		-0.01245	0.26123	
0.425	-0.00378	0.35271		-0.00378	0.25763	
0.450	0.00840	0.35293		0.00840	0.25763	
0.475	0.01357	0.34865		0.01357	0.25425	
0.500	0.02225	0.34580		0.02225	0.24447	
0.525	0.03082	0.34238		0.03082	0.23730	
0.550	0.03820	0.33827		0.03820	0.22872	
0.575	0.04827	0.33375		0.04827	0.22776	
0.600	0.05455	0.32849		0.05455	0.22118	
0.625	0.06562	0.32256		0.06562	0.21461	
0.650	0.07420	0.31541		0.07420	0.20744	
0.675	0.08257	0.30848		0.08257	0.19986	
0.700	0.09165	0.30020		0.09165	0.19488	
0.725	0.10032	0.29407		0.10032	0.18340	
0.750	0.10300	0.28887		0.10300	0.17470	
0.775	0.11767	0.28320		0.11767	0.16549	
0.800	0.12635	0.25642		0.12635	0.15588	
0.825	0.13502	0.24232		0.13502	0.14586	
0.850	0.14220	0.22458		0.14220	0.13486	
0.875	0.15227	0.20951		0.15227	0.12467	
0.900	0.16105	0.19111		0.16105	0.11247	
0.910	0.16652	0.18339		0.16652	0.10888	
0.920	0.17146	0.1728		0.17146	0.09447	
0.940	0.17493	0.15891		0.17493	0.09466	0.09518
0.950	0.17840	0.15037		0.17840	0.08790	0.08799
0.960	0.18187	0.14732		0.18187	0.08790	0.08799
0.970	0.18524	0.13201		0.18524	0.07494	0.09210
0.980	0.18881	0.12424		0.18881	0.07507	0.09722
0.990	0.19228	0.11560		0.19228	0.07016	0.09587
1.000	0.19545	0.10643		0.19545	0.06506	0.10412

PRC4 UTILITY PROGRAM - FLOW AREA CALCULATION

SCME FT FINAL 1B...R.J.ROWLEY..10-06-87...

HOT TO COLD CONVERSION RADII

HOT	COLD
4.12000	4.10981
4.21650	4.20520
4.31300	4.20048
4.40950	4.39574
4.50600	4.49096
4.60250	4.58617
4.69900	4.68142
4.79550	4.77670
4.89200	4.87201
4.98850	4.96745
5.08500	5.06377

RECTAGGER ANGLE DEGREES = 0.0 RADIANS = 0.0
PLATFORM RADII LE ID = 4.12000 LE OD = 5.10000
TE ID = 4.12000 TE OD = 5.10000

GAGING RADII INNER = 4.12000 OUTER = 5.08500

NUMBER OF ELABES FOR GAGING = 50

STAGGER IN DEGREES IS -3.00 TO 3.00 IN INCREMENTS OF 0.50

IN CLACG IS -6 TO 6

TOLER = 0.0

STAGGER (DEGREES)	HOT FLOW (SQ IN)	(CHANGE FLOW AREA (SQ IN)	COLD FLOW (SQ IN)
-3.00000	8.80176	-11.51731	
-2.50000	9.05540	-4.58258	
-2.00000	9.24852	-7.65567	
-1.50000	9.44107	-5.73306	
-1.00000	9.63305	-3.81615	
-0.50000	9.81445	-1.90507	
0.0	10.01525	0.0	9.91182
0.50000	10.10547	1.89112	
1.00000	10.39508	3.79252	
1.50000	10.58406	5.67441	
2.00000	10.77240	7.55993	
2.50000	10.96008	9.43346	
3.00000	11.14713	11.30156	

SSME FT FINAL 1B - R.J. ROWEY - 10-05-87

NOMINAL FLOW AREA - SQ. IN.

HOT - 10.015

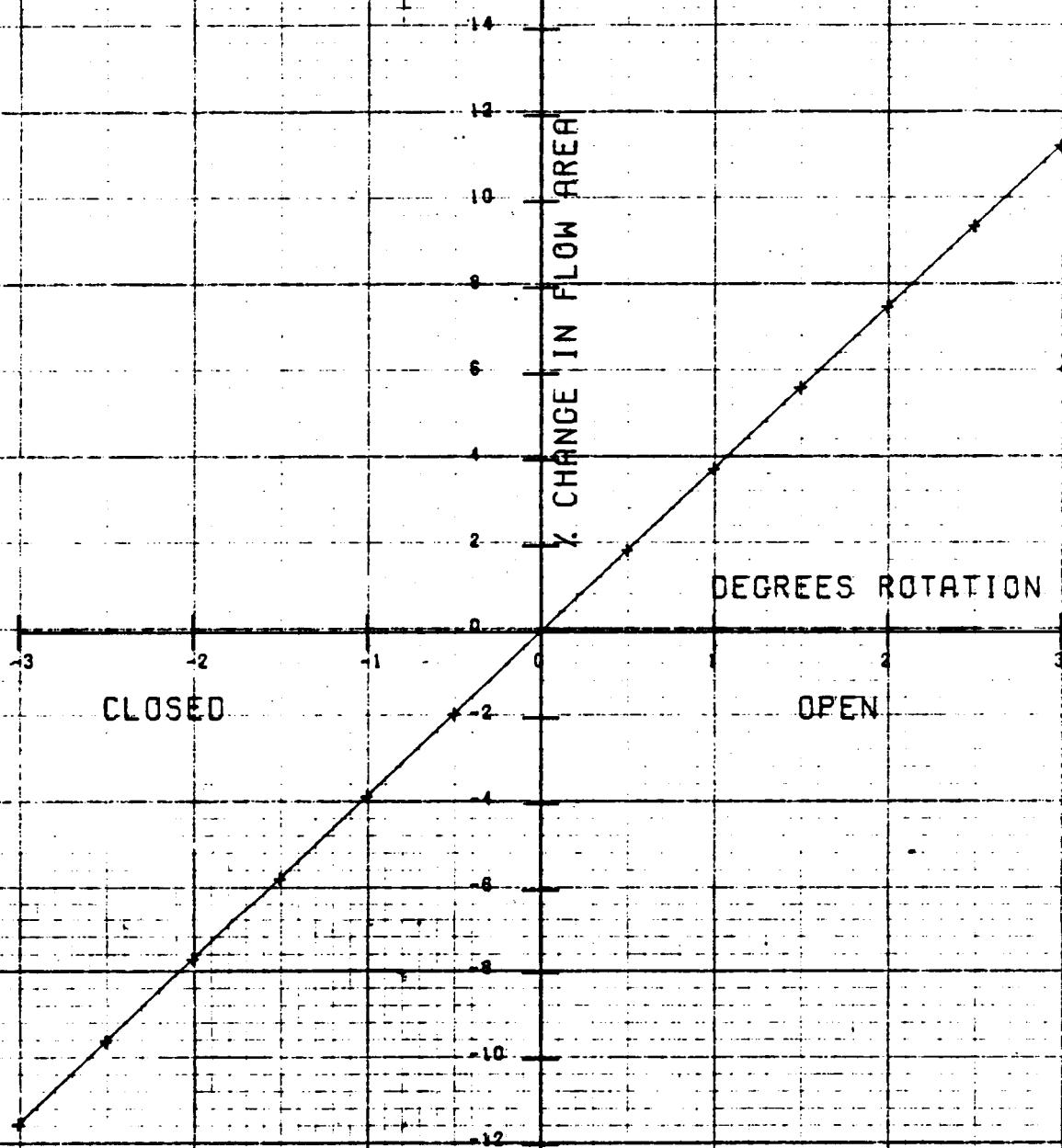
COLD - 9.912

CLOSED

OPEN

% CHANGE IN FLOW AREA

DEGREES ROTATION



SSME FT FILE - 1B...R.J.RONEY..10-06-87...
 OPERATING CONDITION 1 109 PCT...ADP...R.J.RONEY...5-15-87...
 MAE = 157.9 MTR = 153.2 MMH = 38.3 MTR = 0.0

RPM = 36421. MVR = 147.2 MMH = 36.8 MTR = 0.0

%S	RADIUS	SIG P/A	SHRD P/A	LE	TE	CONV
				NET BENDING		
0	4.1200	32523.	0.	-13.	-26.	19.
10	4.2165	31155.	0.	51.	564.	-263.
20	4.3130	29502.	0.	190.	796.	-432.
30	4.4095	27490.	0.	495.	601.	-564.
40	4.5060	25047.	0.	676.	705.	-712.
50	4.6025	22115.	0.	944.	606.	-868.
60	4.6990	18669.	0.	1120.	59.	-978.
70	4.7955	14732.	0.	1094.	695.	-936.
80	4.8920	10328.	0.	803.	370.	-671.
90	4.9885	5442.	0.	326.	158.	-268.
100	5.0850	0.	0.	0.	0.	-0.
%S	RADIUS	XOFF	YOFF	LE	TE	GAS BENDING
				GAS BENDING		
0	4.1200	0.0	0.0	32656.	25800.	-20838.
10	4.2165	0.0071	0.0112	29162.	21226.	-26771.
20	4.3130	0.0156	0.0220	25803.	16127.	-23765.
30	4.4095	0.0252	0.0326	22532.	15315.	-20858.
40	4.5060	0.0355	0.0431	19223.	1253.	-16952.
50	4.6025	0.0460	0.0537	15784.	10225.	-14546.
60	4.6990	0.0563	0.0645	12170.	7630.	-11141.
70	4.7955	0.0661	0.0753	8396.	5193.	-7569.
80	4.8920	0.0753	0.0862	4658.	2889.	-4097.
90	4.9885	0.0859	0.0969	1460.	945.	-1265.
100	5.0850	0.0920	0.1074	0.	0.	-0.
%S	RADIUS	AREA	PULL	LE	TE	SHROUD MISALIGNMENT
				SHROUD MISALIGNMENT		
0	4.1200	0.1137	510.2	0.	0.	0.
10	4.2165	0.1023	469.9	0.	0.	0.
20	4.3130	0.0921	433.0	0.	0.	0.
30	4.4095	0.0831	400.0	0.	0.	0.
40	4.5060	0.0753	370.9	0.	0.	0.
50	4.6025	0.0685	345.4	0.	0.	0.
60	4.6990	0.0626	322.8	0.	0.	0.
70	4.7955	0.0574	302.1	0.	0.	0.
80	4.8920	0.0526	282.0	0.	0.	0.
90	4.9885	0.0481	261.7	0.	0.	0.
100	5.0850	0.0436	0.0	0.	0.	0.
%S	RADIUS	MTR	MMT	MCG	MTC	THETA N
				THETA N		
0	4.1200	-153.3	-147.1	153.2	167.2	-66.03
10	4.2165	-121.4	-122.6	126.1	119.2	-38.39
20	4.3130	-94.3	-97.6	98.0	94.2	-47.53
30	4.4095	-71.4	-75.7	75.1	72.1	-66.29
40	4.5060	-52.1	-52.4	55.1	53.0	78.35
50	4.6025	-36.0	-36.6	38.3	36.8	45.55
60	4.6990	-22.9	-20.8	24.5	23.5	30.52
70	4.7955	-12.7	-10.9	13.8	13.3	25.02
80	4.8920	-5.5	-4.5	6.1	5.9	23.69
90	4.9885	-1.3	-1.0	1.5	1.5	24.32
100	5.0850	0.0	0.0	-0.0	0.0	-8.13

SSME FT FINAL LB...R.J.ROMEY. 10-06-87
 DENSITY = 0.31200 WEIGHTING RADIU INNER = 4.1200 OUTER = 5.0650
 SHROUD VOLUME = 0.0 SHROUD THICKNESS = 0.0
 SHROUD MISALIGNMENT = 0.0 RADIUS OF SHROUD = 0.0
 WEIGHT OF AIRFOIL = 0.02168 WEIGHT OF SHROUD = 0.0
 TOTAL WEIGHT = 1.08415 NUMBER OF BLADES = 50

SUMMARY OF SECTION PROPERTIES

Z/S	RADIUS	AREA	I MIN	I MAX	THETA	XBAR	YBAR
0	4.1200	0.1137	0.1299E-02	0.0096E-02	-14.71	0.0003	0.0567
10	4.2165	0.1023	0.1233E-02	0.6963E-02	-16.38	0.0154	0.0679
20	4.3130	0.0921	0.1130E-02	0.5948E-02	-18.52	0.0239	0.0787
30	4.4095	0.0831	0.1000E-02	0.5057E-02	-21.13	0.0335	0.0893
40	4.5060	0.0753	0.8568E-03	0.4289E-02	-24.19	0.0439	0.0999
50	4.6025	0.0685	0.7038E-03	0.3637E-02	-27.65	0.0543	0.1105
60	4.6990	0.0626	0.5573E-03	0.3090E-02	-31.42	0.0645	0.1212
70	4.7955	0.0576	0.4251E-03	0.2633E-02	-35.40	0.0743	0.1320
80	4.8920	0.0526	0.3128E-03	0.2247E-02	-39.50	0.0835	0.1429
90	4.9885	0.0481	0.2227E-03	0.1920E-02	-43.64	0.0921	0.1536
100	5.0850	0.0436	0.1535E-03	0.1640E-02	-47.80	0.1002	0.1641
Z/S	RADIUS	K	L	MAX T	AX.WIDTH	C1	C2
0	4.1200	0.0	0.2763E-03	0.2746	0.9000	0.1930	0.2229
10	4.2165	0.0	0.2179E-03	0.2939	0.8600	0.2012	0.2173
20	4.3130	0.0	0.1725E-03	0.3109	0.8200	0.2061	0.2083
30	4.4095	0.0	0.1382E-03	0.3256	0.7800	0.2079	0.1962
40	4.5060	0.0	0.1125E-03	0.3379	0.7400	0.2015	0.1817
50	4.6025	0.0	0.9284E-04	0.3483	0.7000	0.2017	0.1654
60	4.6990	0.0	0.7738E-04	0.3563	0.6600	0.1935	0.1483
70	4.7955	0.0	0.6638E-04	0.3635	0.6200	0.1824	0.1311
80	4.8920	0.0	0.5318E-04	0.3698	0.5800	0.1693	0.1146
90	4.9885	0.0	0.4246E-04	0.3765	0.5400	0.1550	0.0992
100	5.0850	0.0	0.3506E-04	0.3844	0.5000	0.1405	0.0850
Z/S	RADIUS	C3	CLE	CTE	C4	ALPHA	B
0	4.1200	0.2086	0.3991	0.4815	0.0045	72.79	0.9244
10	4.2165	0.2130	0.3715	0.4721	0.0351	72.07	0.8846
20	4.3130	0.2145	0.3466	0.4623	0.0456	70.99	0.8669
30	4.4095	0.2125	0.3262	0.4523	0.0586	69.51	0.8115
40	4.5060	0.2067	0.3043	0.4422	0.0734	67.59	0.7790
50	4.6025	0.1969	0.2873	0.4320	0.0688	65.22	0.7498
60	4.6990	0.1857	0.2736	0.4220	0.0837	62.37	0.7246
70	4.7955	0.1678	0.2636	0.4119	0.0967	59.09	0.7036
80	4.8920	0.1503	0.2572	0.4020	0.0884	55.45	0.6869
90	4.9885	0.1326	0.2544	0.3922	0.0969	51.53	0.6743
100	5.0850	0.1150	0.2547	0.3827	0.1032	47.44	0.6654

P824 UTILITY PROGRAM - STRESS CALCULATION

SSME FT FINAL 1B...R.J.ROMEY..10-06-87...
ENGINE OPERATING CONDITION
MAE RPM TITLE
1 157.9 36421. 109 PCT...ADP...R.J.ROMEY...5-15-87...
XGBR = 153.17601 XGBM = 38.29401 XGBT = 0.0
YGBR = 147.18486 YGBM = 36.79601 YGBT = 0.0
DENSITY = 0.31200 WEIGHTING RADIU INNER = 4.1200 CUTER = 5.0850
SHROUD VOLUME = 0.0 SHROUD THICKNESS = 0.0
SHROUD MISALIGNMENT = 0.0 RADIUS OF SHROUD = 0.0

SSME FT FINAL 1B R. J. ROWEY 6-25-87

STRESS VS. SPAN

ARE = LS7.9
RPM = 36421.
PULL = 3898.
CENS = 0.31200

1 SFC P/R SHROUD
2 SFC P/R FOIL SHROUD
3 SFC P/R + SFC SHMIS
4 SFC P/R + SFC SHMIS + SFC OS
5 SFC P/R + SFC SHMIS + SFC NB

90

80

70

60

50

STRESS
KSI

40

30

20

10

0

CONDITION:

EG9 PCT. ADR. R. J.
ROWEY.. 5-15-87.

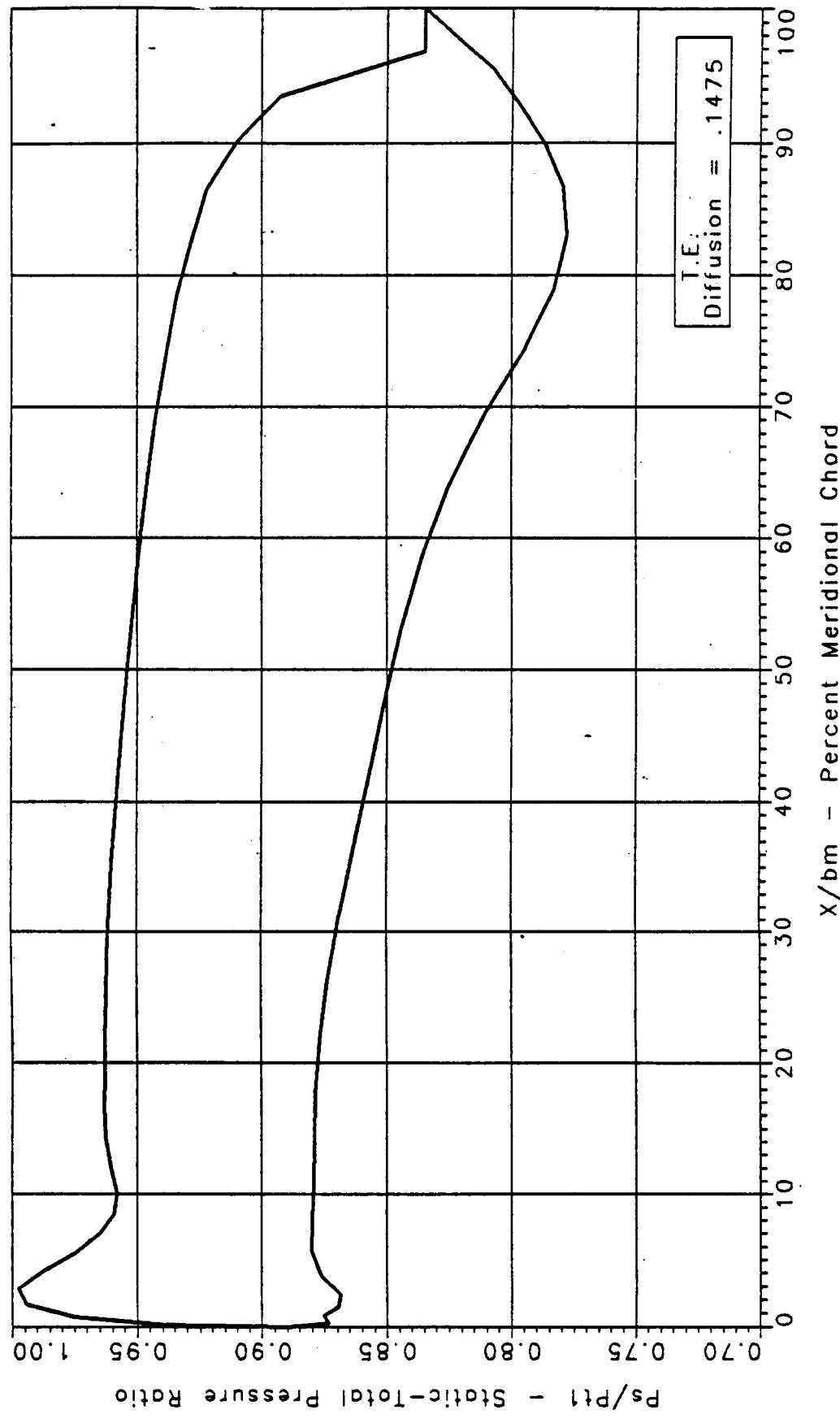
STRESS PT

% SPAN BASED ON
WEIGHT RADII
R10F = 4.1200
R10D = 5.0850

0 10 20 30 40 50 60 70 80 90 100 %S

PRATT & WHITNEY
SSME ATD Fuel Pump Turbine
3-D Pressure Distribution "C" - Mesh
Poisson Generated

First Stage Blade: 0% Span

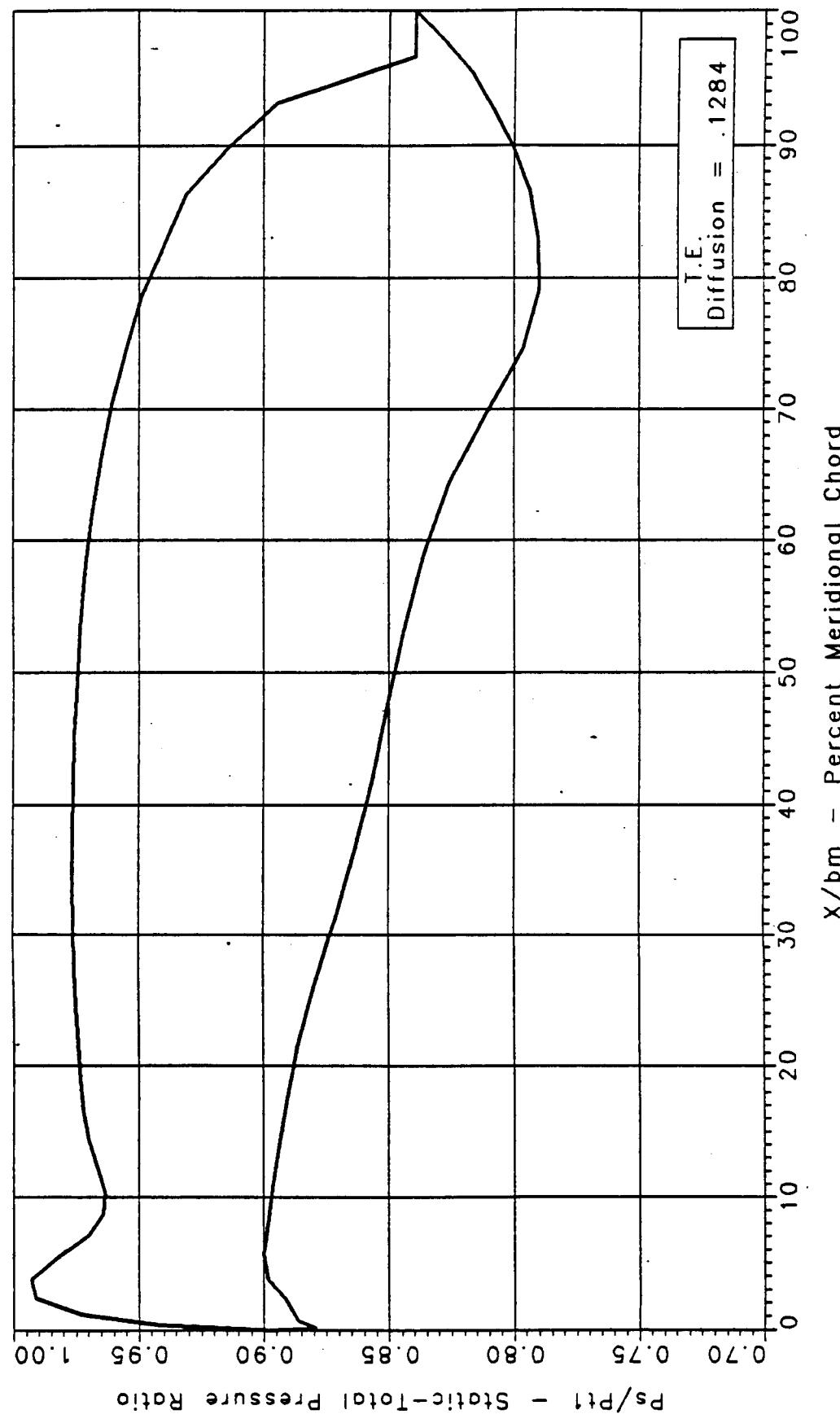


08/20/87
DLS

PRATT & WHITNEY
SSME ATD Fuel Pump Turbine
3-D Pressure Distribution "C"-Mesh
Poisson Generated (V310)

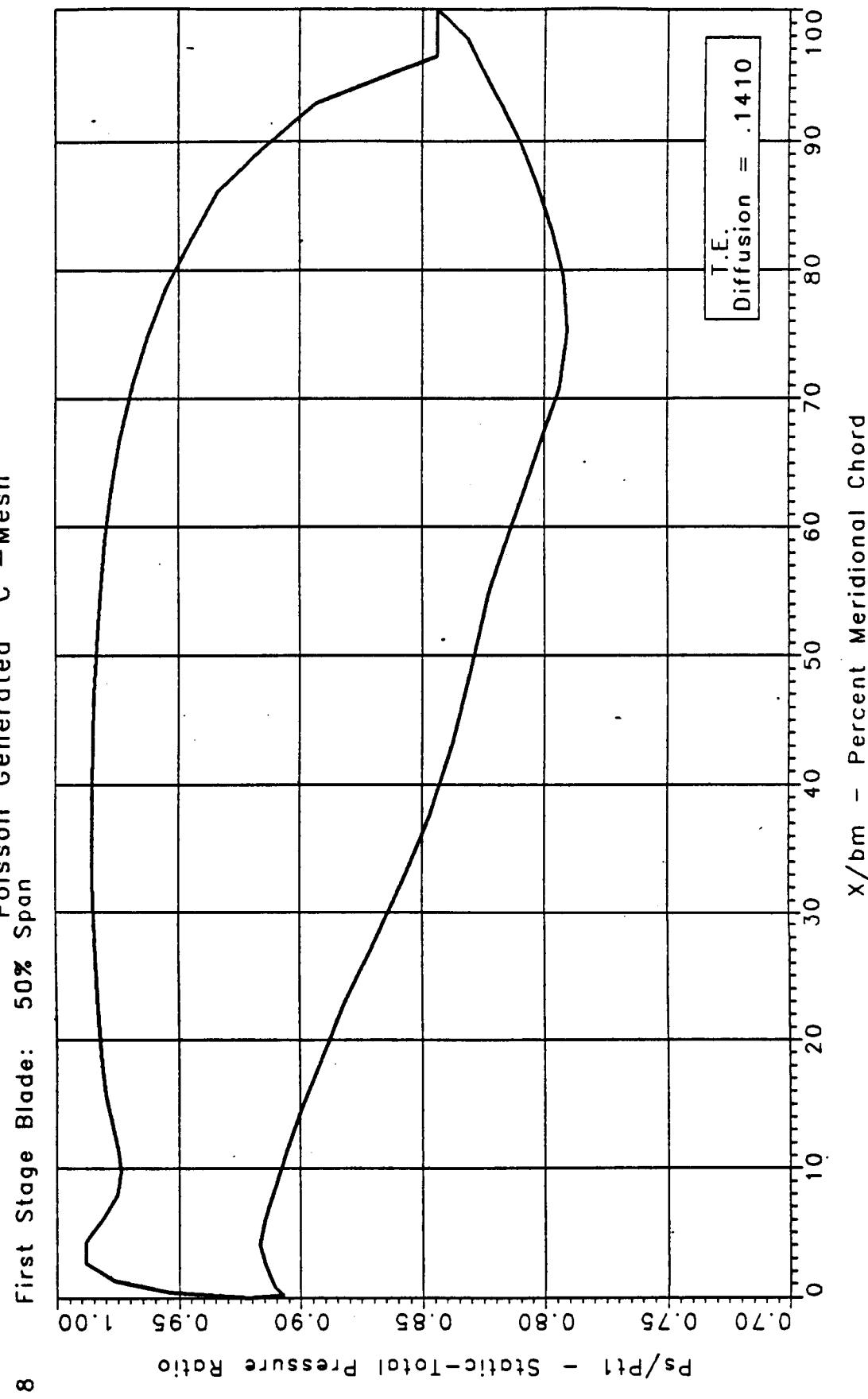
First Stage Blade: 25% Span

7



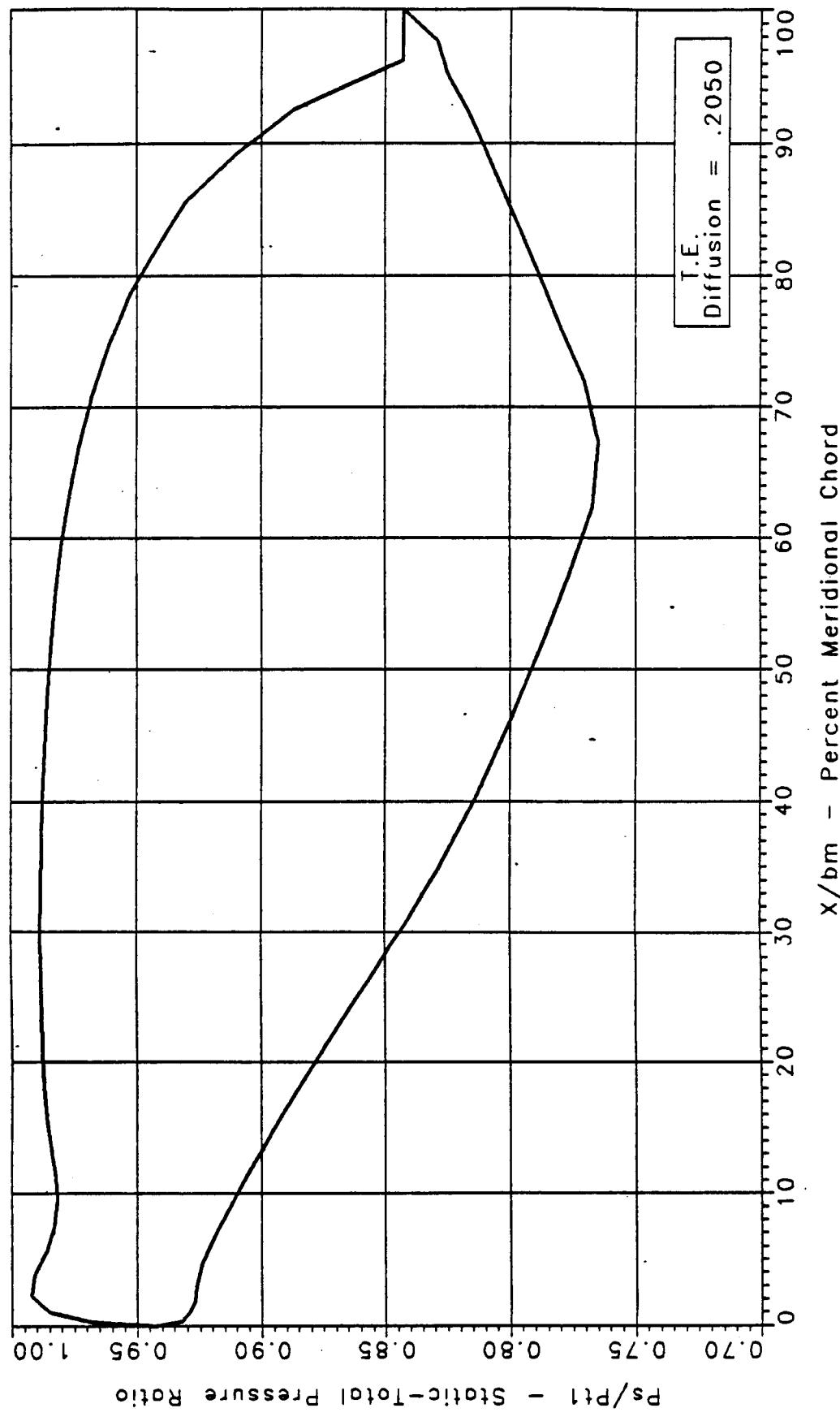
08/20/87
DLS

PRATT & WHITNEY
 SSME ATD Fuel Pump Turbine
 3-D Pressure Distribution "C" - Mesh
 Poisson Generated



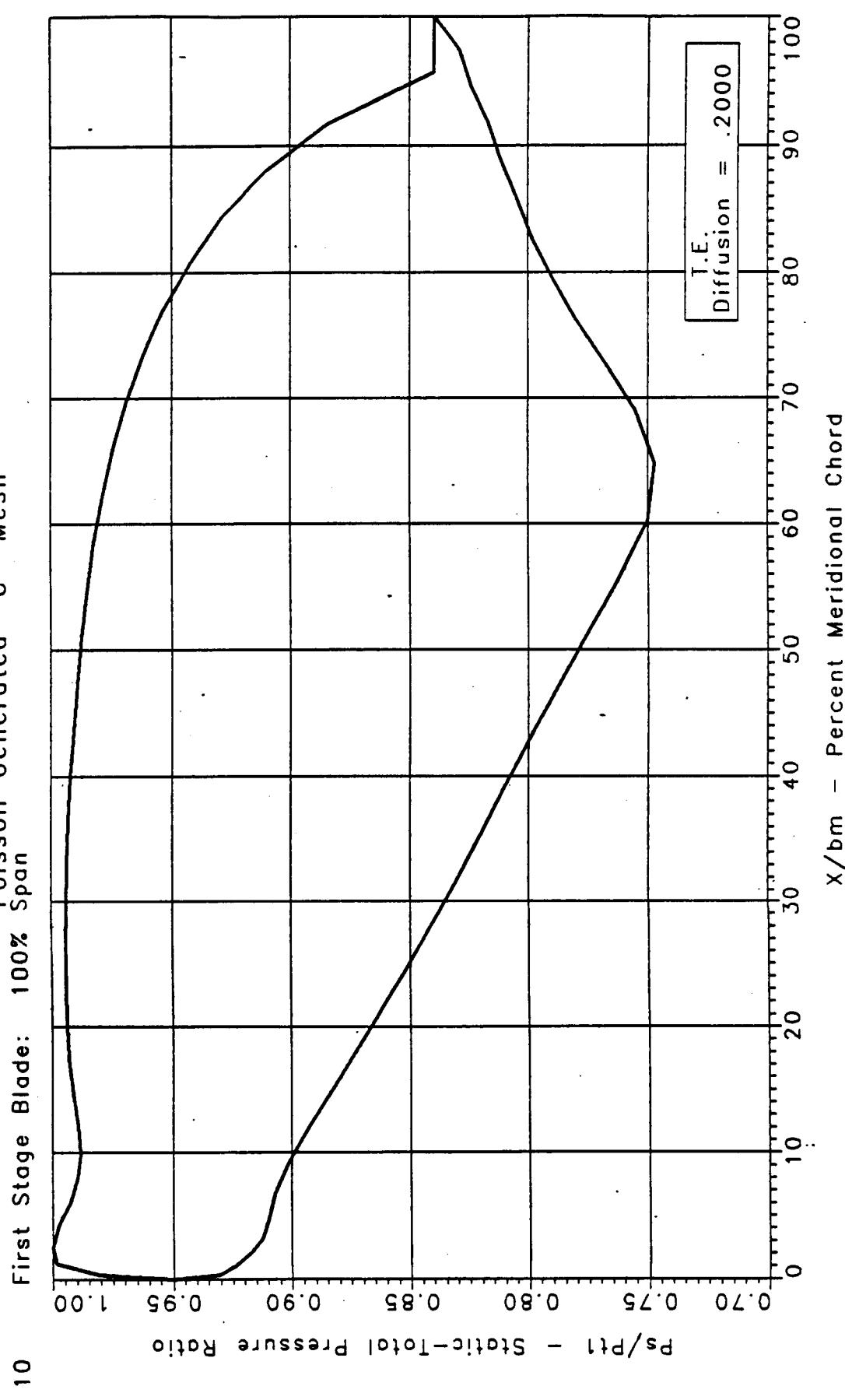
08/20/87
 DLS

PRATT & WHITNEY
 SSME ATD Fuel Pump Turbine
 3-D Pressure Distribution "C" - Mesh
 Poisson Generated
 First Stage Blade: 75% Span



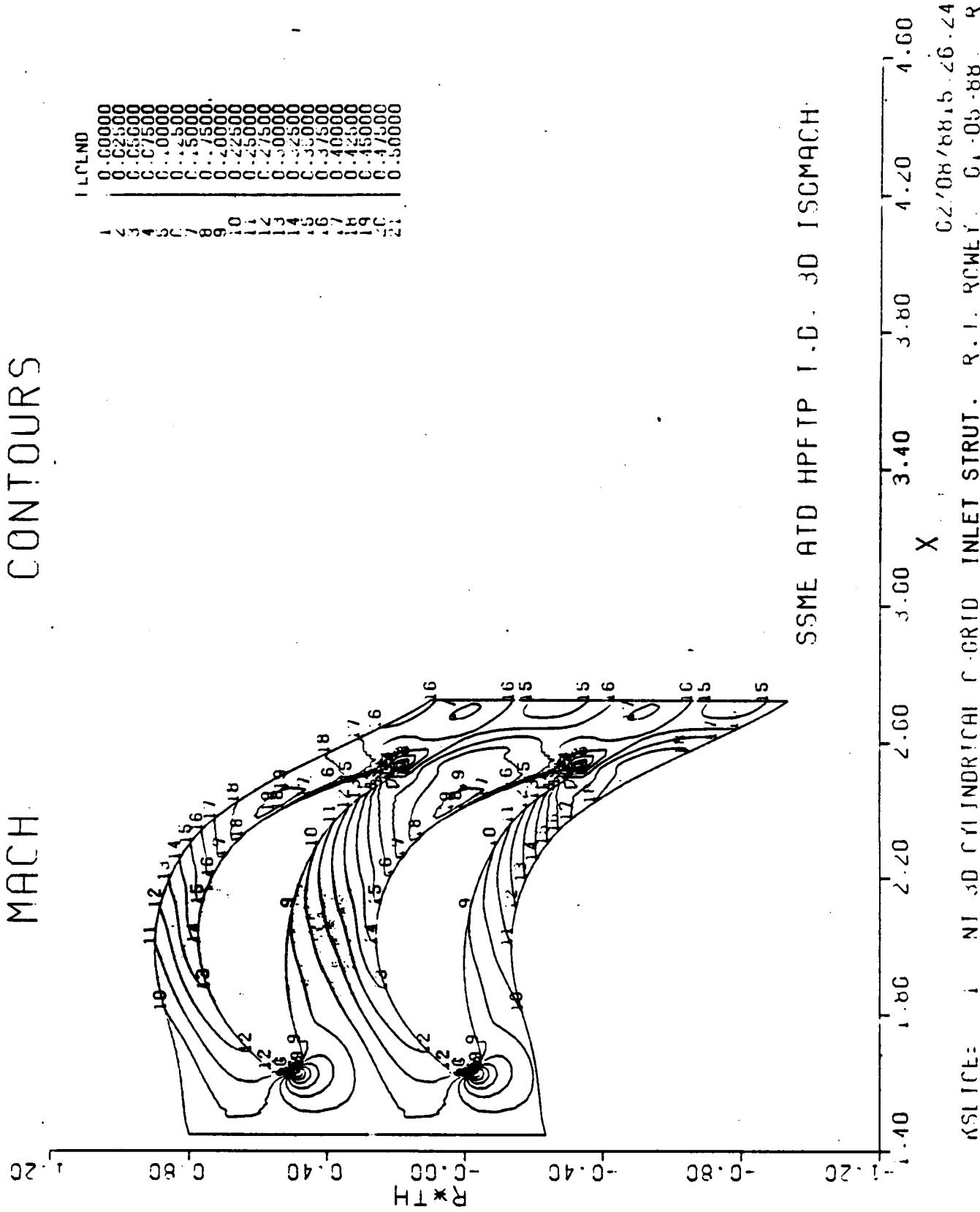
08/20/87
 DL

PRATT & WHITNEY
SSME AID Fuel Pump Turbine
3-D Pressure Distribution "C"-Mesh
Poisson Generated



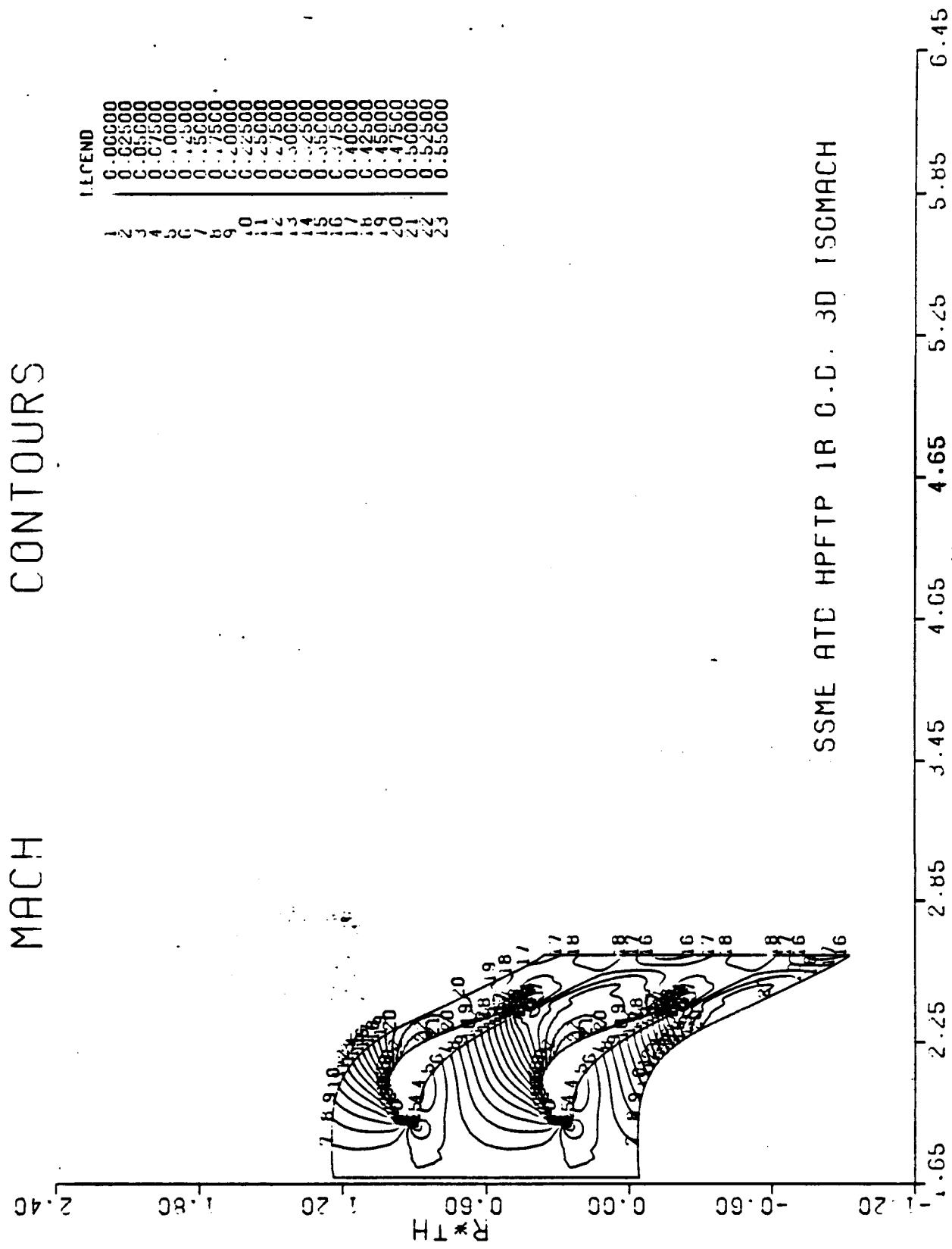
08/20/87
DLS

MACH. CONTOURS



MACH

CONTOURS

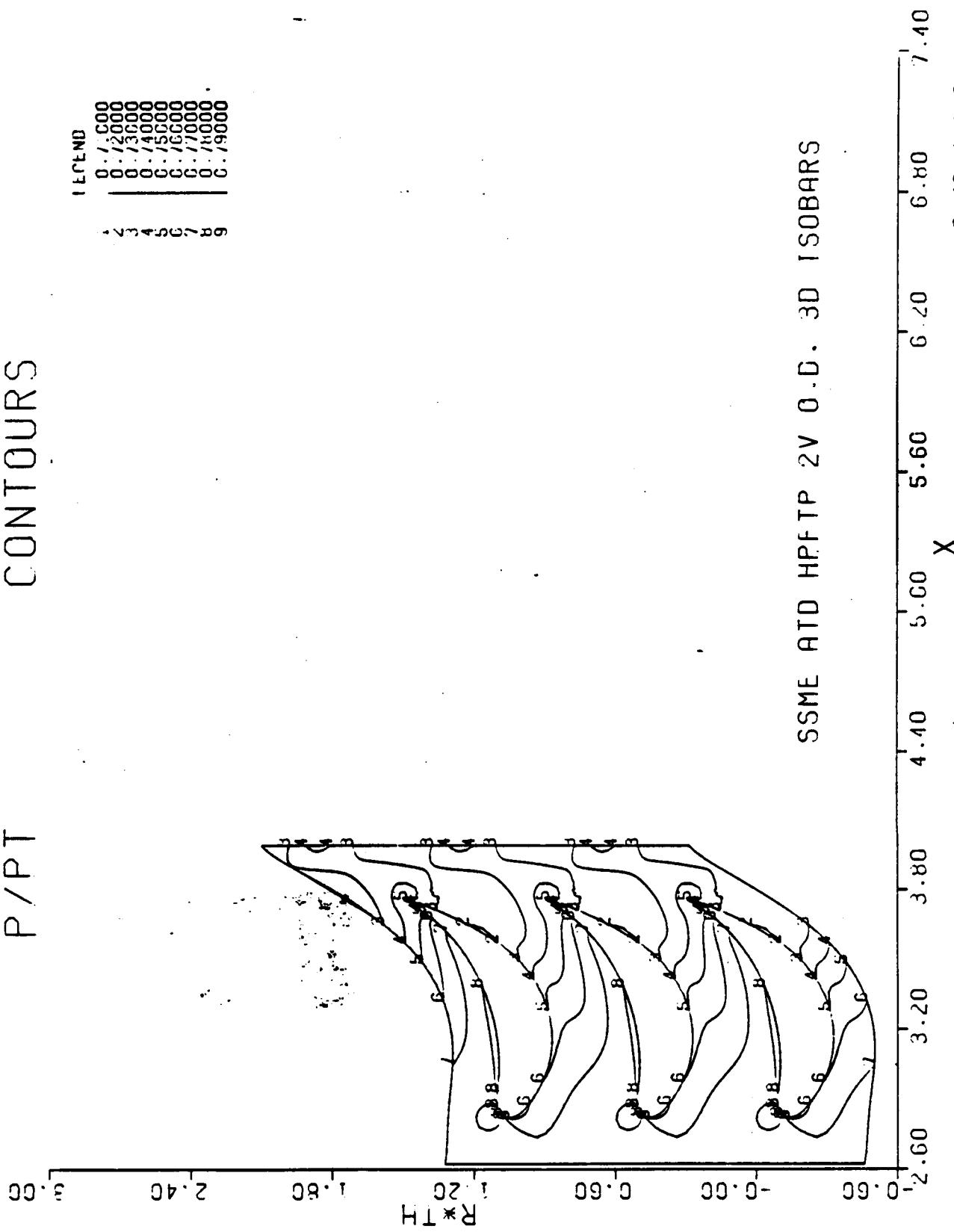


KSI.ICE.: 9 NJ 3D CFD INDR!CRI C-CRDI INLET STRUT.. R. I. ROWER. 02/08/88 16.18.41
R

P / PT

CONTOURS

1 ERTND
0.12000
0.13000
0.14000
0.15000
0.16000
0.17000
0.18000
0.19000
0.20000

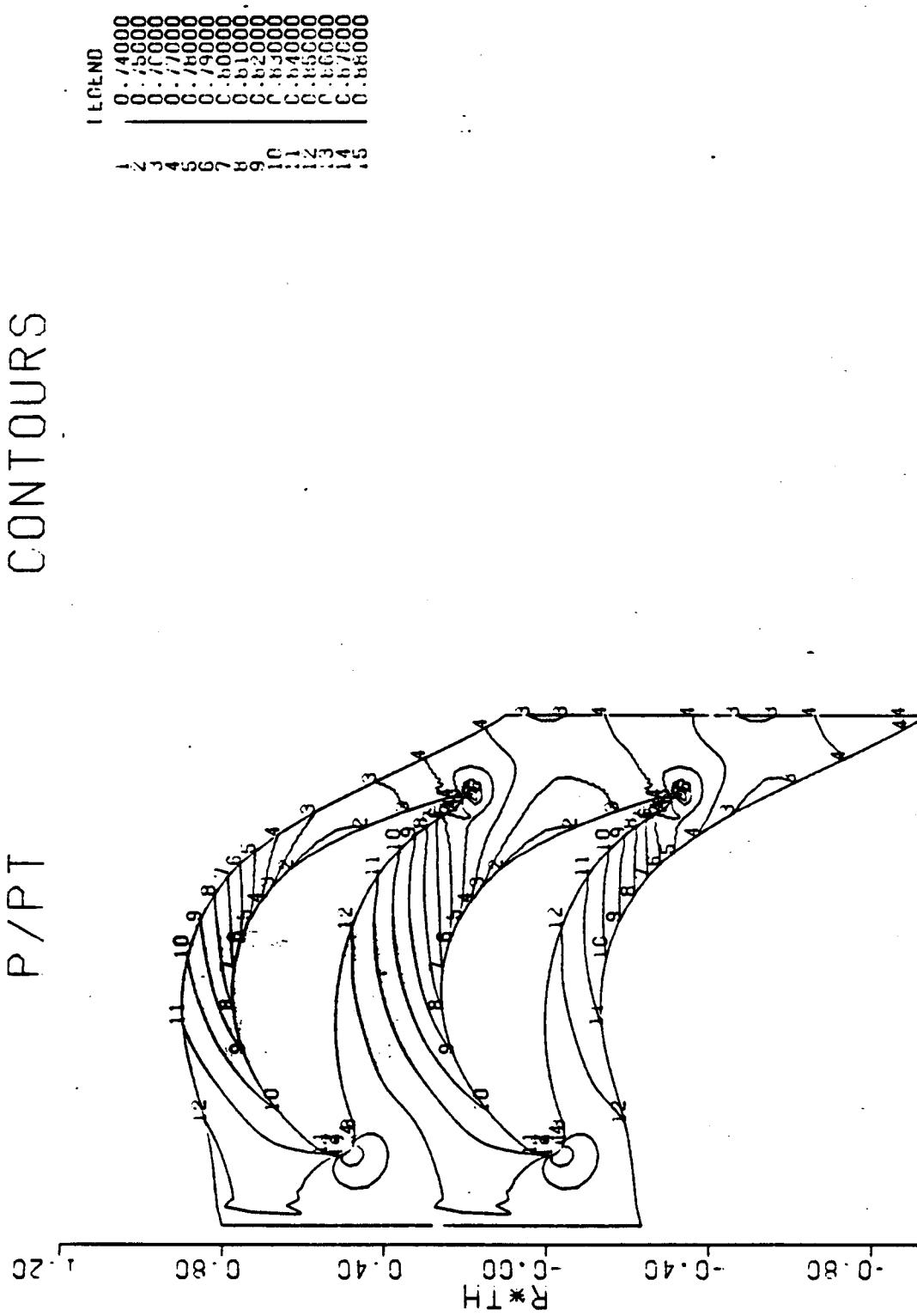


KSLICE: 9 NI 3D CYLINDRICAL C-GRID INLET STRUT.. R.J. RGWEI. C1-05-88. R

02/08/88 16:21:22

P / P_T

CONTOURS



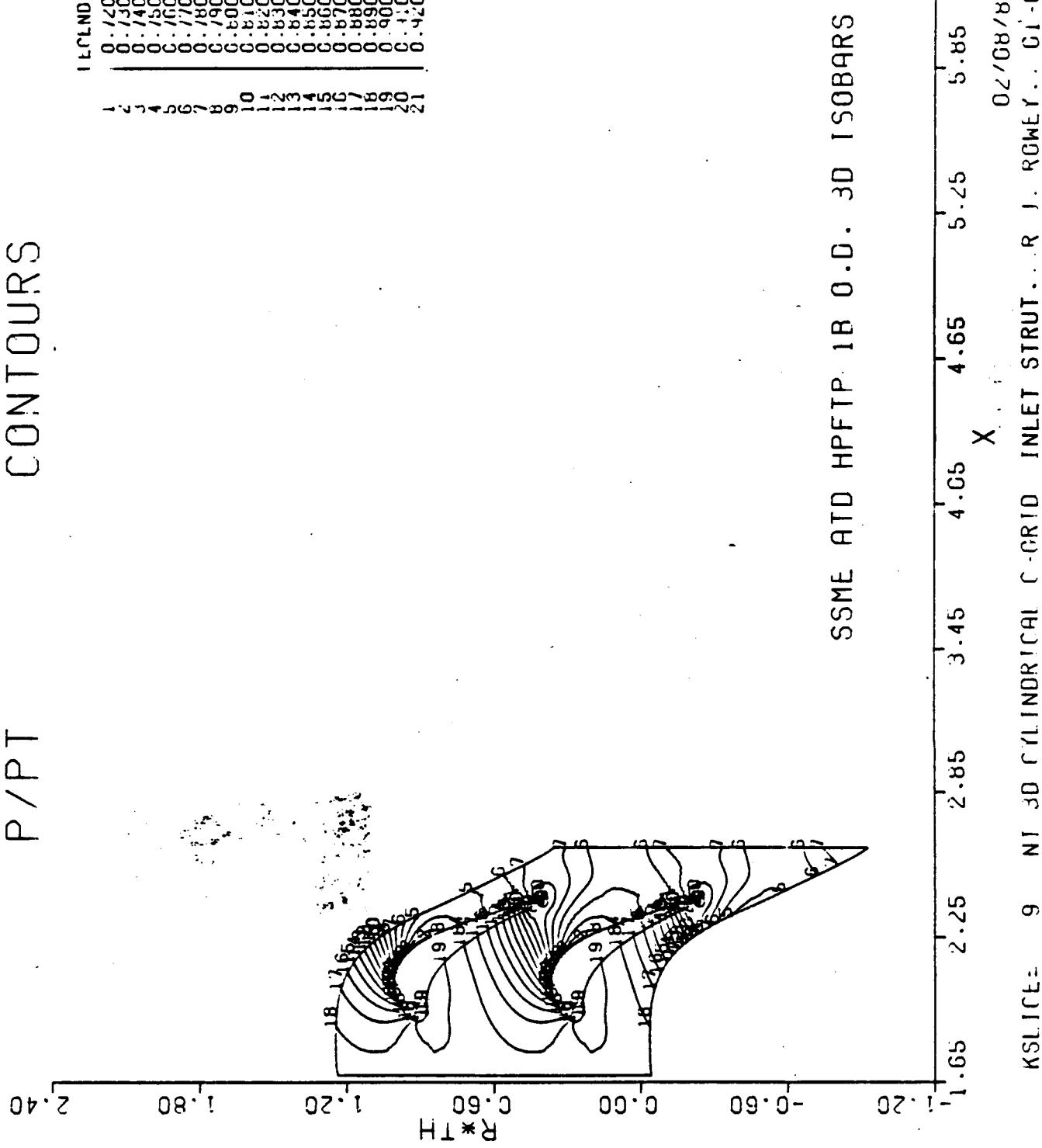
SSME ATD HPFTP 1B 1.D 3D ISCBARS

X
0.40 1.80 2.20 2.60 3.00 3.40 3.80 4.20 4.60
KSI FILE: 1 NI 3D CYLINDRICAL GRID INLET STRUT.. R. I. RGWF.R. G. -05 -88. R
0.20 0.60 0.80 1.20 1.40 1.60 1.80 2.00 2.20 2.40 2.60 2.80 3.00 3.20 3.40 3.60 3.80 4.00 4.20 4.40 4.60

P / PT

CONTOURS

LAND
0 : /4000
0 : /3000
0 : /4000
0 : /5000
0 : /6000
0 : /7000
0 : /8000
0 : /9000
0 : /10000
0 : /11000
0 : /12000
0 : /13000
0 : /14000
0 : /15000
0 : /16000
0 : /17000
0 : /18000
0 : /19000
0 : /20000
1.4 1.5 1.6 1.7 1.8 1.9 1.0 1.1 1.2 1.3 1.4 1.5 1.6 1.7 1.8 1.9 2.0 2.1



U456 - FINITE TRANSITION INTEGRAL BOUNDARY LAYER DECK

DATE 07/29/87

TIME 11:51:11

FIRST BLADE... R. J. ROHEY... 06/11/87... RJ

RFUEL2 U760 RLE= 4.36 RTE= 4.37

4.00% TU

INLET

EXIT

MACH NO.

0.266

0.490

GAS ANGLES

33.63

19.32

SUCTION SIDE

REF. REYNOLDS NO. *****

0.0100

0.0080

0.0080

0.0070

0.0060

0.0050

0.0040

0.0030

0.0020

0.0010

0.0

-0.0

0.5

1.0

1.5

2.0

S DISTANCE (INCHES)

TRANSITION CHART

0.0

0.5

1.0

1.5

2.0

S DISTANCE (INCHES)

U456 - FINITE TRANSITION INTEGRAL BOUNDARY LAYER DECK

DATE 07/29/87

TIME 11:47:26

FIRST BLADE... R. J. ROWEY... 06/11/87... RJ

RFUEL2 U760 RTE= 4.61 RTE= 4.61

4.00% TU

INLET EXIT

MACH NO. 0.233 0.490

GRS ANGLES 36.31 20.04

SUCTION SIDE

REF. REYNOLDS NO. *****

0.0100

0.0080

0.0060

0.0040

0.0020

0.0010

0.0

0.0050

0.0040

0.0030

0.0020

0.0010

0.0

0.0

0.5

1.0

1.5

2.0

S DISTANCE (INCHES)

TRANSITION CHART

S DISTANCE (INCHES)

2.0

U456 - FINITE TRANSITION INTEGRAL BOUNDARY LAYER DECK

DATE 07/28/87 TIME 11:14:46

FIRST BLADE... R. J. ROWEY... 06/11/87... RJ

RFUEL2.UZ60.RLE= 4.85 RTE= 4.86

4.002 TU

INLET

EXIT

MACH NO. 0.207

0.494

GAS RINGLES 48.63

20.58

SUCTION SIDE

REF. REYNOLDS NO. ~~XXXXXXXXXX~~

0.0100

0.0090

0.0080

0.0070

0.0060

0.0050

0.0040

0.0030

0.0020

0.0010

0.0

0.0

0.0

0.5

1.0

1.5

2.0

S DISTANCE (INCHES)

TRANSITION CHART

S DISTANCE (INCHES)

2.0

U456 - FINITE TRANSITION INTEGRAL BOUNDARY LAYER DECK

DATE 07/28/87 TIME 11:30:52

FIRST BLADE... R. J. ROWEY... 06/11/87... RJ

RFUEL2 U760 RLE 5.10 RTE 5.10

4.00% TU

INLET

EXIT

MACH NO. 0.194

0.485

GR5 PINGLES 67.67

23.61

SUCTION SIDE

REF. REYNOLDS NO. 1000000

0.0100

0.0090

0.0080

0.0070

0.0060

0.0050

0.0040

0.0030

0.0020

0.0010

0.0

FAR FWD. DOWN

0.0

0.3

0.5

0.8

1.0

S DISTANCE (INCHES)

TRANSITION CHART

0.0

0.3

0.5

0.8

1.0

S DISTANCE (INCHES)

U456 - FINITE TRANSITION INTEGRAL BOUNDARY LAYER DECK

DATE 07/29/87 TIME 11:42:53

FIRST BLADE...R. J. ROWEY...06/11/87...RJ

RFUEL2 UZ60 RLF= 4.12 RTF= 4.12

4.00% TU

	INLET	EXIT
MACH NO.	0.299	0.498
GAS TANGLES	39.00	21.34

SUCTION SIDE

REF. REYNOLDS NO. *****

0.0100

0.0090

0.0080

0.0070

0.0060

0.0050

0.0040

0.0030

0.0020

0.0010

0.0

0.0

0.0

0.5

1.0

1.5

2.0

S DISTANCE (INCHES)

TRANSITION CHART

0.0

0.5

1.0

1.5

2.0

S DISTANCE (INCHES)

U456 - FINITE TRANSITION INTEGRAL BOUNDARY LAYER DECK

DATE - 07/29/87 - - - TIME - 11:14:45 - - -

FIRST VANE... R. J. RONEY... 06/11/87... RJR

FUEL2 11760. RIF= 4.92 RTF= 4.84

4.00% TU

INLET EXIT

MACH NO. 0.111 0.445

GRD ANGLES 89.95 159.30

PRESSURE SIDE

REF. REYNOLDS NO. *****

0.0100

0.0080

0.0060

0.0070

0.0050

0.0050

0.0045

0.0035

0.0020

0.0010

0.0

EIGHT CMMT

0.0 0.5 1.0 1.5 2.0

S DISTANCE (INCHES)

TRANSITION CHART

0.0 0.5 1.0 1.5 2.0

S DISTANCE (INCHES)

U456 - FINITE TRANSITION INTEGRAL BOUNDARY LAYER DECK

DATE 07/29/87 TIME 11:20:53

FIRST VANE... R. J. ROWNEY... 06/11/87... RJR

FUEL2 U760 RLE 5.20 RTE 5.07

4.00% TU

INLET

EXIT

MACH NO. 0.108 0.422
GAS ANGLES 88.97 154.59

PRESSURE SIDE

REF REYNOLDS NO. *****

0.0100

0.0080

0.0060

0.0040

0.0020

0.0010

0.0000

0.00040

0.00020

0.00010

0.00000

0.0

0.0

0.5

1.0

1.5

2.0

S DISTANCE (INCHES)

TRANSITION CHART

0.0

0.5

1.0

1.5

2.0

S DISTANCE (INCHES)